

Investigation of Riemann solver with internal reconstruction (RSIR) for the Euler equations

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Abstract

The Riemann solver with internal reconstruction (RSIR) of Carmouze et al. (2020) is investigated, revisited and improved for the Euler equations. In this reference, the RSIR approach has been developed to address the numerical resolution of the non-equilibrium two-phase flow model of Saurel et al. (2017). The main idea is to reconstruct two intermediate states from the knowledge of a simple and robust intercell state such as HLL, regardless the number of waves present in the Riemann problem. Such reconstruction improves significantly the accuracy of the HLL solution, preserves robustness and maintains stationary discontinuities. Consequently, when dealing with complex flow models, such as the aforementioned one, RSIR-type solvers are quite easy to derive compared to HLLC-type ones that may require a tedious analysis of the governing equations across the different waves. In the present contribution, the RSIR solver of Carmouze et al. (2020) is investigated, revisited and improved with the help of thermodynamic considerations, making a simple, accurate, robust and positive Riemann solver. It is also demonstrated that the RSIR solver is strictly equivalent to the HLLC solver of Toro et al. (1994) for the Euler equations when the Rankine-Hugoniot relations are used. In that sense, the RSIR approach recovers the HLLC solver in some limit as well as the HLL one in another limit and can be simplified at different levels when complex systems of equations are addressed. For the sake of clarity and simplicity, the derivations are performed in the context of the one-dimensional Euler equations. Comparisons and validations against the conventional HLLC solver and exact solutions are presented.

Keywords: Riemann solver, RSIR, HLL, HLLC, hyperbolic

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1. Introduction

Derivation of an appropriate Riemann solver for complex flow models is a difficult task. Many waves may be present, some eigenvalues may have large multiplicity and some of the equations may be non-conservative. Examples of such models are the MHD equations (Balsara et al. (2012) [1]), compressible solid-fluid models (Gavrilyuk et al. (2008) [2]), non-equilibrium two-phase flow models such as the one of Saurel et al. (2017) [3], as well as the Godunov-Peshkov-Romenski (GPR) model (Peshkov et al. (2019) [4]). Also, when dealing with material interfaces, large density jumps may be present and volume fraction as well as density positivity is mandatory. Same requirement is needed for high speed flows, where vacuum conditions may appear. It seems that the most appropriate Riemann solver in these conditions is the HLLC solver of Toro et al. (1994) [5]. However, for specific models, its derivation may be non-trivial [6], [7].

In Carmouze et al. (2020) [8] an alternative is given and a new Riemann solver with internal reconstruction (RSIR) is designed. It relies on the following observation. It is usually quite easy to derive a single intermediate state solver, such as Rusanov (1961) [9] or HLL (Harten et al. (1983) [10]) even for complicated flow models. These solvers are very robust and positive but too dissipative for transport and stationary contact waves. Extensions of the HLL solver to include more wave information have been developed in Einfeldt et al. (1991) [11], Toro et al. (1994) [5], Linde (2002) [12] and Dumbser and Balsara (2016) [13].

In Carmouze et al. (2020) [8] the single intermediate solution is used to rebuild two intermediate states, thanks to an additional relation. These two intermediate states contain in most situations enough information to improve significantly accuracy and preserve robustness.

The contribution of Carmouze et al. (2020) [8] was mainly motivated by the numerical approximation of the non-equilibrium two-phase flow model of Saurel et al. (2017) [3] that involves a series of theoretical challenges as it is hyperbolic degenerate, presents non-conservative terms and exhibits non self-similar solutions. The complexity of the corresponding model prompted the authors to develop a solver based on internal reconstruction of intermediate states, computed from a simple and robust intercell state.

Thanks to the RSIR approach, stationary interfaces are maintained and numerical dissipation is reduced while circumventing the difficulties related to the construction of a HLLC-type solver. Chiapolino and Saurel (2020) [14] provide illustrative results of the RSIR solver in this two-phase flow context.

In Carmouze et al. (2020) [8], thermodynamic considerations have been introduced in the RSIR solver for the Euler equations and a similar treatment has been developed to address numerical resolution of Saurel et al.'s (2017) [3] model.

The underlying philosophy of this approach relies on the assumption that most of the physics is present in the two extreme waves and only one contact wave, that has to be identified. If the contact wave cannot be defined clearly, the method becomes irrelevant. But it seems that in most flow models such as the Euler equations and the above-mentioned models, identification of the contact wave is possible and often easy.

The remaining waves, when present, are captured by the scheme during computations even if they are omitted in the Riemann problem. This is the same philosophy as the Rusanov and HLL

solvers, except that an extra intermediate wave is added, preserving stationary discontinuities. The basic concept of the RSIR approach relies on three ingredients:

- 45 – A simple and robust intermediate solution state such as HLL or Rusanov;
- A consistency relation resulting from the integral form of conservation laws in a control volume containing a three-wave structure of a Riemann problem (two extreme waves and one contact wave);
- 50 – A heuristic relation linking the jumps of the two intermediate states across the contact wave.

The consistency equation is essential for the RSIR approach and is well-established for conservation laws. It links the simple and robust but single-state HLL (or Rusanov) solution and the two intermediate solution states. Its practical use requires nonetheless the heuristic relation that offers multiple choices depending on the properties of the governing equations.

55 In the present paper, the heuristic relation is addressed through thermodynamic considerations. This approach has been introduced in Carmouze et al. (2020) [8]. It is here investigated, revisited and improved. Second, it is demonstrated that the RSIR limit recovers the HLLC solver for the Euler equations when the heuristic relation is addressed through the Rankine-Hugoniot relations. Computed results confirm this observation. In that sense, the RSIR approach recovers
60 the HLLC solver in some limit as well as the HLL one in another limit and can be simplified at different levels when complex systems of equations are addressed.

For the sake of clarity and simplicity, only the one-dimensional Euler equations are considered for the construction of the RSIR solver. Comparisons with the conventional HLLC solver and exact solutions are then available. The aim is to illustrate the simplicity, accuracy, robustness
65 and flexibility of the RSIR solver.

This paper is organized as follows. The 1D Euler equations are briefly introduced in Section 2 along with the HLL and HLLC solvers as they correspond to the two limits of the RSIR solver. The RSIR solver is then addressed in Section 3. Its discrete entropy production is addressed in Section 4. The RSIR limit is then examined in Section 5 showing that the HLLC solver
70 is recovered. Numerical results are provided in Section 6. They show that the RSIR solver provides results in close agreement with those of the HLLC solver while being relatively simple and flexible. Conclusions are finally given in Section 7.

2. The HLL and HLLC Riemann solvers for the Euler equations

The 1D Euler equations of compressible fluids consist of a system of conservation laws,

$$\frac{\partial \mathbf{U}}{\partial t} + \frac{\partial \mathbf{F}}{\partial x} = 0, \tag{2.1}$$

75 where $\mathbf{U} = (\rho, \rho u, \rho E)^T$ is the vector of conservative variables and $\mathbf{F} = (\rho u, \rho u^2 + p, (\rho E + p) u)^T$ the corresponding flux vector. The notations are conventional in fluid dynamics, ρ denotes the density, u the velocity and $E = e + \frac{1}{2}u^2$ the total energy. The pressure p is given by a convex

equation of state (EOS), as a function for example of internal energy e and density ρ . The ideal-gas EOS is used in the present contribution as $p(\rho, e) = (\gamma - 1)\rho e$ where $\gamma = 1.4$ represents the specific heat ratio. This EOS is retained for the sake of conciseness only. The numerical method addressed in the present work is valid for any convex EOS. This system is strictly hyperbolic with wave speeds $\lambda_1 = u$, $\lambda_2 = u - c$ and $\lambda_3 = u + c$. The sound speed is defined as $c = \sqrt{\left(\frac{\partial p}{\partial \rho}\right)_s}$ where s denotes the entropy. In the context of the ideal-gas EOS, the sound speed reads $c(p, \rho) = \sqrt{\frac{\gamma p}{\rho}}$.

The RSIR solver is based on the HLL (Harten et al. (1983) [10]) approximate solution, or its simplified version due to Rusanov (1961) [9]. It aims to construct a robust Riemann solver with limited dissipation, producing results similar to those provided by the HLLC solver.

The HLL and HLLC Riemann solvers are briefly recalled in the following. As will be seen further, the solutions provided by those two approximate Riemann solvers are precisely the two limits of the RSIR solver. It is consequently necessary to introduce the HLL and HLLC solvers beforehand. For a detailed presentation, the reader is referred to Toro's textbook (2009) [15], particularly Chapter 10 of this reference.

HLL and HLLC Riemann solvers

The approximate HLL solver requires estimates for two extreme waves emerging from the initial discontinuity. It results from the integration of the corresponding equations over a two-wave Riemann problem.

A more accurate method is the HLLC solver, developed by Toro et al. (1994) [5]. This method assumes a three-wave model for the structure of the Riemann problem, resulting in better resolution of intermediate waves. The two wave structures (HLL and HLLC) of the Riemann problem are depicted in Fig. 1.

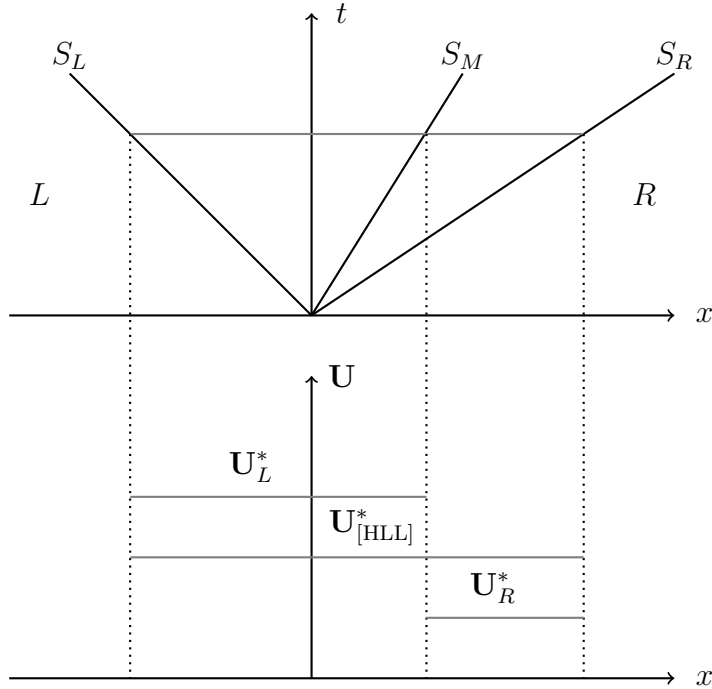


Figure 1: Schematic representation in the (x, t) diagram of the HLL, HLLC and RSIR solvers. The HLL solver assumes a two-wave structure (S_L and S_R) while the intermediate wave S_M is restored with the HLLC solver. The RSIR solver is based on the HLL single state solution complemented by a consistency relation reconstructing two intermediate states \mathbf{U}_L^* , \mathbf{U}_R^* and a heuristic relation linking the jump of the two intermediate states across the contact wave S_M .

100 The central idea of the HLL solver is to assume a wave configuration that consists of two waves separating three constant states. The extreme waves denoted S_L and S_R are estimated following Davis (1988) [16],

$$S_L = \min(u_L - c_L, u_R - c_R) \quad \text{and} \quad S_R = \max(u_L + c_L, u_R + c_R), \quad (2.2)$$

where indexes L and R denote the left and right states at a given cell boundary. Note that a subsonic wave pattern is assumed in the present work, *i.e.* $S_L < 0$ and $S_R > 0$. These simple wave speed estimates yield accurate results. Moreover, they are convenient for complicated EOS and more sophisticated models than the Euler equations.

Both HLL and HLLC consider waves as discontinuities. Related jump conditions are the well-known Rankine-Hugoniot (RH) conditions:

$$\mathbf{F}_k^* = \mathbf{F}_k + S_k (\mathbf{U}_k^* - \mathbf{U}_k), \quad k = L, R, \quad (2.3)$$

110 where S_k denotes the speed of the considered wave k . Note that the states involved in relation (2.3) are spatial integral averages. So strictly speaking these are not the classical Rankine-Hugoniot conditions connecting limiting values left and right of a discontinuity but rather the ‘‘Averaged Rankine-Hugoniot’’ conditions. However the specification ‘‘Averaged’’ will be omitted

in the rest of the paper.

In the HLL solver, no distinction is made between states \mathbf{U}_R^* and \mathbf{U}_L^* . The solution state in
 115 the HLL approximation reads,

$$\mathbf{U}_{[\text{HLL}]^*} = \frac{\mathbf{F}_R - \mathbf{F}_L + S_L \mathbf{U}_L - S_R \mathbf{U}_R}{S_L - S_R}. \quad (2.4)$$

The resulting HLL Riemann solver forms the basis of very efficient and robust approximate
 Godunov-type methods. However, as the intermediate wave is omitted, the HLL solver produces
 more dissipation than the HLLC one. This is not problematic for fast flows as discontinuity are
 captured and in general maintained sharp enough during sufficiently long time, but becomes
 120 problematic for slower flows. Particularly, the HLL solver is unable to maintain discontinuities
 at rest.

The HLLC scheme is a modification of the HLL scheme, whereby the missing contact in the
 Euler equations is restored. The solutions for the two intermediate state vectors \mathbf{U}_L^* and \mathbf{U}_R^* are
 sought. Similarly to the HLL solver, where the Rankine-Hugoniot relations are used across the
 125 two extreme waves, the same jump conditions are used across the intermediate wave yielding
 contact discontinuity conditions:

$$\begin{cases} p_L^* = p_R^* = p^*, \\ u_L^* = u_R^* = u^* = S_M. \end{cases} \quad (2.5)$$

As the extreme waves S_L and S_R are known from (2.2), algebraic manipulations of the mass and
 momentum Rankine-Hugoniot relations (2.3) provide the pressure solutions in the left and right
 perturbed states,

$$p_k^* = p_k + \rho_k (S_k - u_k) (S_M - u_k), \quad k = L, R. \quad (2.6)$$

130 The equality of the pressures allows determination of the intermediate speed S_M as a function
 of speeds S_L and S_R , namely,

$$S_M = u_L^* = u_R^* = \frac{p_R - p_L + (\rho u)_L (S_L - u_L) - (\rho u)_R (S_R - u_R)}{\rho_L (S_L - u_L) - \rho_R (S_R - u_R)} = \frac{U_{[\text{HLL}]^*, \text{momentum}}}{U_{[\text{HLL}]^*, \text{mass}}}. \quad (2.7)$$

The two intermediate solution states are computed with the help of the Rankine-Hugoniot
 relations (2.3) and the corresponding values p_L^* and p_R^* . The solutions can be written concisely
 as,

$$\mathbf{U}_{k[\text{HLLC}]^*} = \frac{S_k \mathbf{U}_k - \mathbf{F}_k + p_k^* \mathbf{D}}{S_k - S_M} \quad \text{with} \quad \mathbf{D} = [0, 1, S_M]^T, \quad k = L, R. \quad (2.8)$$

135 The numerical fluxes, solution of the Riemann problem, are provided by Eq. (2.3) according to
 the Rankine-Hugoniot relations across the left S_L or right S_R wave depending on the sign of the
 contact wave S_M speed.

Unlike the HLL solver, the HLLC solver is able to maintain stationary discontinuities. How-

140 ever, the construction of a HLLC-type solver may be quite difficult for more sophisticated systems such as two-phase flow models or magnetohydrodynamics equations where many waves are present in the Riemann problem.

Recently, the authors developed in Carmouze et al. (2020) [8] a Riemann solver with internal reconstruction (RSIR). It is based on the HLL single state solution, a consistency relation reconstructing two intermediate states \mathbf{U}_L^* , \mathbf{U}_R^* and a heuristic relation linking the jump of the two intermediate states across the contact wave S_M . In the present paper, the heuristic relation 145 is revisited and improved.

3. Riemann solver with internal reconstruction (RSIR)

The RSIR solver is based on internal reconstruction of intermediate states, computed from a simple and robust intercell state, such as Rusanov (1961) [9] or HLL (Harten et al. (1983) [10]). 150 The HLL solution state $\mathbf{U}_{[\text{HLL}]}$ is known from Eq. (2.4) and the contact wave speed S_M is also known from the HLL solution (Eq. (2.7)). The extreme waves S_L and S_R are known with the help of Davis' estimates (2.2).

The aim is now to construct two intermediate states \mathbf{U}_L^* and \mathbf{U}_R^* as illustrated in Fig. 1. The average HLL state and the two intermediate solution states are linked through the consistency 155 relation (illustrated in Fig. 1),

$$(S_R - S_L) \mathbf{U}_{[\text{HLL}]}^* = (S_R - S_M) \mathbf{U}_R^* + (S_M - S_L) \mathbf{U}_L^*. \quad (3.1)$$

This consistency relation can be rewritten as,

$$\mathbf{U}_{[\text{HLL}]}^* = \omega_R \mathbf{U}_R^* + \omega_L \mathbf{U}_L^*, \quad (3.2)$$

with

$$\omega_R = \frac{S_R - S_M}{S_R - S_L} \quad \text{and} \quad \omega_L = \frac{S_M - S_L}{S_R - S_L}. \quad (3.3)$$

Relation (3.2) involves two unknown states, \mathbf{U}_L^* and \mathbf{U}_R^* . Consequently an extra relation, linking the jumps of the two intermediate states across the contact wave, is needed.

160 First let us introduce the heuristic of Linde (2002) [12],

$$\mathbf{U}_R^* - \mathbf{U}_L^* = \beta (\mathbf{U}_R - \mathbf{U}_L). \quad (3.4)$$

In this relation, β represents a viscosity parameter. When β is taken equal to zero, the HLL approximation is recovered. When $\beta = 1$ the reconstruction tends to the HLLC representation but is not equivalent, as interface conditions (2.5) are ignored in Linde's approach.

Relation (3.4) is then combined with Relation (3.2) resulting in,

$$\begin{cases} \mathbf{U}_{L[\text{Linde}]}^* = \mathbf{U}_{[\text{HLL}]}^* - \omega_R \beta (\mathbf{U}_R - \mathbf{U}_L), \\ \mathbf{U}_{R[\text{Linde}]}^* = \mathbf{U}_{[\text{HLL}]}^* + \omega_L \beta (\mathbf{U}_R - \mathbf{U}_L). \end{cases} \quad (3.5)$$

165 With the help of Relation (3.5), corresponding fluxes are computed thanks to the RH relations

(2.3). As shown in Carmouze et al. (2020) [8], Eq. (3.5) sometimes yields correct results but large spurious oscillations may appear as well depending on the initial conditions.

This observation motivated reconsideration of the heuristic relation, now based on thermodynamic considerations, resulting in significant improvements and yielding robust and accurate solutions. The heuristic relation now reads,

$$\mathbf{U}_R^* - \mathbf{U}_L^* = \Psi, \quad (3.6)$$

where Ψ is the jump vector linking the two intermediate states \mathbf{U}_L^* and \mathbf{U}_R^* . As previously, Relation (3.6) is combined with Eq. (3.2) resulting in,

$$\begin{cases} \mathbf{U}_{L[\text{RSIR}]}^* = \mathbf{U}_{[\text{HLL}]}^* - \omega_R \Psi, \\ \mathbf{U}_{R[\text{RSIR}]}^* = \mathbf{U}_{[\text{HLL}]}^* + \omega_L \Psi. \end{cases} \quad (3.7)$$

Relation (3.7) is the cornerstone of the RSIR approach. It results from the combination of the consistency relation and the jump vector linking the two intermediate states. Equation (3.7) provides the solution of the two intermediate states, built upon the HLL solution. However, its practical use requires knowledge of the jump vector Ψ . This task offers multiple choices depending on the properties of the governing equations.

It is important to note that only an approximate jump vector Ψ is needed to improve the HLL solution through Eq. (3.7). As will be seen further, the different components of the jump vector may sometimes be expressed directly, for example $(\rho_R^* - \rho_L^*)$, or two approximate variables may be obtained separately, for example ρ_R^* and ρ_L^* , giving the jump by subtracting the two quantities. However, in the latest case, approximate variables such as ρ_R^* and ρ_L^* are not to be used as solutions but only as an approximation of the jump Ψ , to be combined with the consistency relation.

Regardless the way the jump vector Ψ is computed, the two solution states \mathbf{U}_L^* and \mathbf{U}_R^* are determined through Eq. (3.7). The solutions are then built upon the HLL one and the approximation of the jump vector.

In the following, crude knowledge of the governing equations' properties is assumed in the aim to address complex flow models. The aim is then to approximate the jump vector Ψ from relations as simple and general as possible while involving relevant physics. As will be seen later in Section 5, when more details of the governing equations are available, the RSIR solver recovers the HLLC one through the introduction of the Rankine-Hugoniot relations in the jump vector Ψ .

The approximation of the jump vector Ψ derived hereafter is based on two ingredients:

- Quasi-isentropic or barotropic variations across right- and left-facing waves;
- Insertion of interface conditions across the contact wave.

Although not strictly correct, in particular across strong shocks, thermodynamic evolutions through right- and left-facing waves are approximated as quasi-isentropic.

Mass relation

200 Thermodynamic evolutions across the right- and left-facing waves are approximated through sound speed definition $c^2 = \left(\frac{\partial p}{\partial \rho}\right)_s$ as,

$$\bar{c}^2 = \frac{p_L^* - p_L}{\rho_L^* - \rho_L} = \frac{p_R^* - p_R}{\rho_R^* - \rho_R}. \quad (3.8)$$

Such approximation has been used for example in Appendix A of [17] to approximate isentropes in tank boundary conditions, when dealing with sophisticated equations of state. The average square sound speed \bar{c}^2 is estimated as,

$$\bar{c}^2 = \max [c_L^2, c_R^2]. \quad (3.9)$$

205 As will be seen in Section 4, this estimate guarantees discrete entropy production. Note that this is different from the original RSIR solver introduced in Carmouze et al. (2020) [8] where the average sound speed is computed with the help of a rough trapezoidal approximation, $\bar{c} = \frac{1}{2}(c_L + c_R)$. Separate sound speeds c_L and c_R have been considered as well instead of a single average one such as (3.9). However, robustness issues appeared with these estimates.

210 Across the contact wave S_M , the interface pressure condition reads,

$$p_L^* = p_R^* = p^*. \quad (3.10)$$

Thus Eq. (3.8) becomes,

$$\begin{cases} p^* = p_R + \bar{c}^2 (\rho_R^* - \rho_R), \\ p^* = p_L + \bar{c}^2 (\rho_L^* - \rho_L). \end{cases} \quad (3.11)$$

Taking the difference of these two relations, the following one is obtained:

$$\rho_R^* - \rho_L^* = \rho_R - \rho_L - \frac{p_R - p_L}{\bar{c}^2}. \quad (3.12)$$

Relation (3.12) corresponds to a modification of the first component of Linde's approximation (3.4). To maintain flexibility of the reconstruction method, parameter β is reintroduced as,

$$\rho_R^* - \rho_L^* = \beta \left(\rho_R - \rho_L + \frac{p_L - p_R}{\bar{c}^2} \right) = \Psi^{mass}. \quad (3.13)$$

215 Parameter β seems convenient to control numerical viscosity, for example to remove the contact wave when dealing with computations in extreme conditions. In all examples considered in the present paper, $\beta = 1$ (unless stated otherwise), corresponding to the least dissipative version. The analysis of β as a viscosity parameter is addressed in Appendix B. With the present version of the RSIR solver, only $\beta = 0$ and $\beta = 1$ are admissible values.

220 Equation (3.13) is an approximation of the mass jump between the two intermediate states. The solution densities are now determined with the help of the mass jump and the consistency

relation (3.7),

$$\begin{cases} \rho_L^* = \rho_{[\text{HLL}]}^* - \omega_R \beta \left(\rho_R - \rho_L + \frac{p_L - p_R}{\bar{c}^2} \right), \\ \rho_R^* = \rho_{[\text{HLL}]}^* + \underbrace{\omega_L \beta \left(\rho_R - \rho_L + \frac{p_L - p_R}{\bar{c}^2} \right)}_{\Psi^{mass}}. \end{cases} \quad (3.14)$$

Momentum relation

Similar relations are deduced to approximate the momentum jump across the contact wave. As $u_L^* = u_R^* = S_M$, Eq. (3.13) implies the following relation,

$$(\rho u)_R^* - (\rho u)_L^* = \beta \left(\rho_R - \rho_L + \frac{p_L - p_R}{\bar{c}^2} \right) S_M = \Psi^{\text{momentum}}. \quad (3.15)$$

The solution momenta consequently read,

$$\begin{cases} (\rho u)_L^* = (\rho u)_{[\text{HLL}]}^* - \omega_R \beta \left(\rho_R - \rho_L + \frac{p_L - p_R}{\bar{c}^2} \right) S_M, \\ (\rho u)_R^* = (\rho u)_{[\text{HLL}]}^* + \underbrace{\omega_L \beta \left(\rho_R - \rho_L + \frac{p_L - p_R}{\bar{c}^2} \right) S_M}_{\Psi^{\text{momentum}}}. \end{cases} \quad (3.16)$$

Energy relation

As quasi-isentropic variations across right- and left-facing waves are supposed, the approximation of the energy jump is determined with the help of the first law of thermodynamics, expressed under the form of Gibbs' relation,

$$de = T \underbrace{ds}_{=0} - p dv, \quad (3.17)$$

where T denotes the temperature, s the entropy and $v = 1/\rho$ the specific volume. Thanks to the quasi-isentropic assumption, Gibbs' relation is approximated as,

$$\begin{cases} e_L^* = e_L - \bar{p}_L (v_L^* - v_L), \\ e_R^* = e_R - \bar{p}_R (v_R^* - v_R), \end{cases} \quad (3.18)$$

where $v_L^* = 1/\rho_L^*$ and $v_R^* = 1/\rho_R^*$ are known from Eq. (3.14). In Relation (3.18), \bar{p}_k is an estimate of the average pressure across the corresponding wave. Multiple choices are available at this level. The most obvious ones are $\bar{p}_k = p_k$ with $k = L, R$ or $\bar{p}_k = p^*$ or a combination of them $\bar{p}_k = \frac{p_k + p^*}{2}$. As will be seen further, the analysis of the entropy production reports that only $\bar{p}_k = p^*$ is an admissible estimate. This estimate is determined from the average square sound speed as,

$$p^* = p_L + \bar{c}^2 (\rho_L^* - \rho_L) = p_R + \bar{c}^2 (\rho_R^* - \rho_R). \quad (3.19)$$

This is another difference with the original RSIR solver introduced in Carmouze et al. (2020) [8] where the EOS $e_k^*(p_k^*, \rho_k^*)$ is used to compute directly the energy jump.

In the present paper, the quasi-isentropic assumption is approximated through both sound speed definition and Gibbs' relation. The total energy jump is finally expressed thanks to Eqs. (3.14) and (3.18),

$$(\rho E)_R^* - (\rho E)_L^* = \beta \left[\rho_R^* \left(e_R^* + \frac{1}{2} S_M^2 \right) - \rho_L^* \left(e_L^* + \frac{1}{2} S_M^2 \right) \right] = \Psi^{energy}, \quad (3.20)$$

where the parameter β is reintroduced for the sake of generality. The solutions for total energies are now built upon the HLL ones as,

$$\begin{cases} (\rho E)_L^* = (\rho E)_{[HLL]}^* - \omega_R \beta \left[\rho_R^* \left(e_R^* + \frac{1}{2} S_M^2 \right) - \rho_L^* \left(e_L^* + \frac{1}{2} S_M^2 \right) \right], \\ (\rho E)_R^* = (\rho E)_{[HLL]}^* + \omega_L \beta \underbrace{\left[\rho_R^* \left(e_R^* + \frac{1}{2} S_M^2 \right) - \rho_L^* \left(e_L^* + \frac{1}{2} S_M^2 \right) \right]}_{\Psi^{energy}}. \end{cases} \quad (3.21)$$

The intermediate solution states are then fully determined for the Euler equations. They are computed upon the HLL solution with the help of the consistency relation and the jump vector,

$$\begin{cases} \mathbf{U}_{L[RSIR]}^* = \mathbf{U}_{[HLL]}^* - \omega_R \Psi, \\ \mathbf{U}_{R[RSIR]}^* = \mathbf{U}_{[HLL]}^* + \omega_L \Psi. \end{cases} \quad (3.22)$$

Solution fluxes

Once states \mathbf{U}_L^* and \mathbf{U}_R^* are determined through (3.22) the various fluxes are computed through the Rankine-Hugoniot relations, according to the sign of S_M ,

$$\begin{cases} \mathbf{F}_R^* = \mathbf{F}_R + S_R (\mathbf{U}_R^* - \mathbf{U}_R), \\ \mathbf{F}_L^* = \mathbf{F}_L + S_L (\mathbf{U}_L^* - \mathbf{U}_L). \end{cases} \quad (3.23)$$

The Riemann solver thus consists of Eq. (2.4) to compute the simple and robust HLL state, Eqs. (3.13), (3.15), (3.20) to approximate the jumps across the intermediate wave S_M and Eq. (3.22) to construct the two intermediate states. Note that the solver does not require explicit formulation of the EOS.

4. Entropy production

The discrete entropy production of the RSIR solver is now addressed. During the construction of the jump vector Ψ and particularly the energy component Ψ^{energy} , the following thermodynamic path is used,

$$de = -\bar{p}dv, \quad (4.1)$$

where \bar{p} is an estimate of the average pressure.

260 The choice of \bar{p} is dictated by the second law of thermodynamics. Let us insert the present thermodynamic path into Gibbs' relation,

$$Tds = de + pdv = -\bar{p}dv + pdv. \quad (4.2)$$

The integration of the previous relation over the unperturbed state ($_{k=L,R}$) and the solution state (*_k) reads,

$$\int_k^{^*_k} Tds = \int_k^{^*_k} (p - \bar{p}) dv. \quad (4.3)$$

265 The right-hand side of Relation (4.3) represents the discrete entropy production across the extreme waves S_L and S_R . With the RSIR solver, the pressure evolves across the extreme waves as,

$$p(\rho) = p_k + \bar{c}^2 (\rho - \rho_k). \quad (4.4)$$

Equation (4.3) consequently transforms to,

$$\int_k^{^*_k} Tds = \int_k^{^*_k} (p_k + \bar{c}^2 (\rho - \rho_k) - \bar{p}) dv. \quad (4.5)$$

The discrete entropy production is satisfied if the right hand-side of Eq. (4.5) is non-negative.

Let us now introduce the pressure estimate as $\bar{p} = p^*$. Relation (4.5) becomes,

$$\int_k^{^*_k} Tds = \int_k^{^*_k} (p_k - p^*) dv + \bar{c}^2 \int_k^{^*_k} \left(\frac{1}{v} - \frac{1}{v_k} \right) dv = (p_k - p^*) (v_k^* - v_k) + \bar{c}^2 \left(\ln \left(\frac{v_k^*}{v_k} \right) - \frac{v_k^*}{v_k} + 1 \right). \quad (4.6)$$

270 However, the RSIR star pressure reads $p^* = p_k + \bar{c}^2 \left(\frac{1}{v_k^*} - \frac{1}{v_k} \right)$. The preceding relation consequently transforms to,

$$\int_k^{^*_k} Tds = \bar{c}^2 \left(\frac{(v_k^* - v_k)^2}{v_k^* v_k} + \ln \left(\frac{v_k^*}{v_k} \right) - \frac{v_k^*}{v_k} + 1 \right). \quad (4.7)$$

For the sake of simplicity, let us introduce $X = \frac{v_k^*}{v_k}$. The discrete entropy production is then satisfied if,

$$f(X) = \frac{(X - 1)^2}{X} + \ln(X) - X + 1 \geq 0. \quad (4.8)$$

275 A simple mathematical function analysis shows that $f(X) \geq 0 \forall X \in]0, +\infty[$ with a minimum corresponding to $f(X = 1) = 0$.

The estimate $\bar{p} = p^*$ consequently satisfies the discrete entropy production making the RSIR solver entropy preserving. It is interesting to emphasize that the amount of discrete entropy is quantified by the averaged sound speed \bar{c}^2 . For the sake of robustness, it is preferable to use the

largest value of \bar{c}^2 . For this reason, the average sound speed is estimated as,

$$\bar{c}^2 = \max [c_L^2, c_R^2], \quad (4.9)$$

as seen in Section 3.

It is also important to emphasize that the previous calculations lead to $f(X) \leq 0 \forall X \in]0, +\infty[$ if $\bar{p} = p_k$ and $f(X) \geq 0$ for $X \in]0, 1]$ if $\bar{p} = \frac{p_k + p^*}{2}$. Those estimates are consequently rejected.

Note that the latest result is not surprising as it indicates that the discrete entropy production is satisfied only for $X = \frac{v_k^*}{v_k} = \frac{\rho_k}{\rho_k^*} < 1$ corresponding to a compression. Indeed the introduction of $\bar{p} = \frac{p_k + p^*}{2}$ into Gibbs' relation $de = -\bar{p}dv$ leads to $e_k^* - e_k + \frac{p_k + p^*}{2}(v_k^* - v_k) = 0$ which is precisely the Hugoniot adiabat, valid for compressions only.

It consequently appears that $\bar{p} = p^*$ (computed with the help of Eq. (3.19)) is the only admissible estimate among the ones examined. Numerical experiments with the present RSIR solver are provided in Section 6 showing efficiency and robustness. However, before analyzing the corresponding results, let us address the limit of the RSIR solver. When more details of the equations are used, meaning they are available, the jump vector Ψ can be approximated with the help of the Rankine-Hugoniot relations. This task is addressed hereafter showing that the HLLC limit is recovered.

5. HLLC limit of the RSIR solver

The RSIR solution relies on Eq. (3.7), recalled hereafter,

$$\begin{cases} \mathbf{U}_{L[\text{RSIR}]}^* = \mathbf{U}_{[\text{HLL}]}^* - \omega_R \Psi, \\ \mathbf{U}_{R[\text{RSIR}]}^* = \mathbf{U}_{[\text{HLL}]}^* + \omega_L \Psi. \end{cases} \quad (5.1)$$

Unlike Section 3 where quasi-isentropic evolutions are assumed, the construction of the jump vector Ψ is now addressed through the set of Rankine-Hugoniot relations.

Mass relation

The Rankine-Hugoniot mass relations read,

$$\begin{cases} \rho_L^* (S_M - S_L) = \rho_L (u_L - S_L), \\ \rho_R^* (S_M - S_R) = \rho_R (u_R - S_R). \end{cases} \quad (5.2)$$

The following jump relation consequently arises,

$$\rho_R^* - \rho_L^* = \rho_R \frac{u_R - S_R}{S_M - S_R} - \rho_L \frac{u_L - S_L}{S_M - S_L}. \quad (5.3)$$

Parameter β is introduced as before for the sake of generality,

$$\rho_R^* - \rho_L^* = \beta \left(\rho_R \frac{u_R - S_R}{S_M - S_R} - \rho_L \frac{u_L - S_L}{S_M - S_L} \right) = \Psi^{\text{mass}}. \quad (5.4)$$

Momentum relation

Thanks to the previous mass jump and the interface condition across the intermediate wave S_M , the momentum jump directly reads,

$$(\rho u)_R^* - (\rho u)_L^* = \beta \left(\rho_R \frac{u_R - S_R}{S_M - S_R} - \rho_L \frac{u_L - S_L}{S_M - S_L} \right) S_M = \Psi^{\text{momentum}}. \quad (5.5)$$

Energy relation

Combination of the mass, momentum and energy Rankine-Hugoniot relations provides after some algebraic manipulations,

$$\begin{cases} (\rho E)_L^* = \frac{(\rho E)_L (u_L - S_L) + p_L u_L - p_L^* S_M}{S_M - S_L}, \\ (\rho E)_R^* = \frac{(\rho E)_R (u_R - S_R) + p_R u_R - p_R^* S_M}{S_M - S_R}, \end{cases} \quad (5.6)$$

where the star pressure is provided by the HLL solution, Eq. (2.6). With the help of these relations, a jump relation across the contact wave is obtained,

$$(\rho E)_R^* - (\rho E)_L^* = \beta \left(\frac{(\rho E)_R (u_R - S_R) + p_R u_R - p_R^* S_M}{S_M - S_R} - \frac{(\rho E)_L (u_L - S_L) + p_L u_L - p_L^* S_M}{S_M - S_L} \right) = \Psi^{\text{energy}}. \quad (5.7)$$

The jump vector Ψ across the contact wave is now approximated with the help of Rankine-Hugoniot relations. An interesting consequence appears as detailed in Appendix B. Parameter β is no longer restricted to values 0 and 1 but belongs to the interval $0 \leq \beta \leq 1$.

The solution vectors (5.1) are then to be compared to the HLLC solutions. Let us first recall the solutions of HLLC solver for the Euler equations (see [15] for details),

$$\mathbf{U}_{k[\text{HLLC}]}^* = \frac{S_k \mathbf{U}_k - \mathbf{F}_k + p_k^* \mathbf{D}}{S_k - S_M} \quad \text{with} \quad \mathbf{D} = [0, 1, S_M]^T, \quad k = L, R \quad (5.8)$$

with

$$p_k^* = p_k + \rho_k (S_k - u_k) (S_M - u_k), \quad (5.9)$$

and

$$S_M = u_L^* = u_R^* = \frac{p_R - p_L + (\rho u)_L (S_L - u_L) - (\rho u)_R (S_R - u_R)}{\rho_L (S_L - u_L) - \rho_R (S_R - u_R)} = \frac{U_{[\text{HLL}]}^{*,\text{momentum}}}{U_{[\text{HLL}]}^{*,\text{mass}}}. \quad (5.10)$$

With the RSIR solver, solution states are given by,

$$\begin{cases} \mathbf{U}_{L[\text{RSIR}]}^* = \mathbf{U}_{[\text{HLL}]}^* - \omega_R \Psi, \\ \mathbf{U}_{R[\text{RSIR}]}^* = \mathbf{U}_{[\text{HLL}]}^* + \omega_L \Psi. \end{cases} \quad (5.11)$$

Examination of the RSIR solution state vector $\mathbf{U}_{L[\text{RSIR}]}^*$ is now addressed,

320 *Mass solution*

The mass solution reads,

$$\rho_{L[\text{RSIR}]}^* = \underbrace{\left(\frac{\rho_R u_R - \rho_L u_L + S_L \rho_L - S_R \rho_R}{S_L - S_R} \right)}_{\rho_{[\text{HLL}]}} - \underbrace{\left(\frac{S_R - S_M}{S_R - S_L} \right)}_{\omega_R} \underbrace{\left(\rho_R \frac{u_R - S_R}{S_M - S_R} - \rho_L \frac{u_L - S_L}{S_M - S_L} \right)}_{\Psi^{\text{mass}}}. \quad (5.12)$$

Equation (5.12) may be written as,

$$\rho_{L[\text{RSIR}]}^* = \frac{\rho_R (u_R - S_R)}{S_L - S_R} - \frac{\rho_L (u_L - S_L)}{S_L - S_R} - \beta \frac{\rho_R (u_R - S_R)}{S_L - S_R} + \beta \left(\frac{S_R - S_M}{S_R - S_L} \right) \rho_L \frac{u_L - S_L}{S_M - S_L}. \quad (5.13)$$

When $\beta = 0$, the HLL solution is recovered. When $\beta = 1$, Eq. (5.13) reduces to,

$$\rho_{L[\text{RSIR}]}^* = \frac{\rho_L (u_L - S_L)}{S_M - S_L}, \quad (5.14)$$

325 which is precisely $\rho_{L[\text{HLLC}]}^*$ according to Eq. (5.8). The density solution is consequently merged with the HLLC one.

Momentum solution

Combination of Eqs. (5.8) and (5.9) gives for the momentum,

$$(\rho u)_{L[\text{HLLC}]}^* = \frac{\rho_L (u_L - S_L)}{S_M - S_L} S_M = \rho_{L[\text{HLLC}]}^* S_M. \quad (5.15)$$

RSIR formula is now expanded for comparison with the HLLC solution,

$$\begin{aligned} (\rho u)_{L[\text{RSIR}]}^* &= \underbrace{\frac{(\rho u)_R (u_R - S_R) - (\rho u)_L (u_L - S_L) + p_R - p_L}{S_L - S_R}}_{(\rho u)_{[\text{HLL}]}} \\ &\quad - \underbrace{\left(\frac{S_R - S_M}{S_R - S_L} \right)}_{\omega_R} \underbrace{\left(\rho_R \frac{u_R - S_R}{S_M - S_R} - \rho_L \frac{u_L - S_L}{S_M - S_L} \right)}_{\Psi^{\text{momentum}}} S_M. \end{aligned} \quad (5.16)$$

After some simplifications it becomes,

$$(\rho u)_{L[\text{RSIR}]}^* = \left(\frac{\rho_R (u_R - S_R)}{S_L - S_R} \right) (u_R + \beta S_M) - \left(\frac{\rho_L (u_L - S_L)}{S_L - S_R} \right) \left(u_L + \beta S_M \frac{S_R - S_M}{S_M - S_L} \right) + \frac{p_R - p_L}{S_L - S_R}. \quad (5.17)$$

330 The last term expresses as,

$$\frac{p_R - p_L}{S_L - S_R} = \frac{\rho_L (S_L - u_L) (S_M - u_L) + \rho_R (S_R - u_R) (u_R - S_M)}{S_L - S_R}, \quad (5.18)$$

determined from the contact wave speed estimate formula (5.10). The RSIR star left momentum jump relation now becomes,

$$(\rho u)_L^*_{[\text{RSIR}]} = \frac{\rho_R (u_R - S_R)}{S_L - S_R} \left[u_R - \beta S_M - (u_R - S_M) \right] - \frac{\rho_L (u_L - S_L)}{S_L - S_R} \left(u_L + \beta S_M \frac{S_R - S_M}{S_M - S_L} + (S_M - u_L) \right). \quad (5.19)$$

We now introduce $\beta = 1$,

$$\begin{aligned} (\rho u)_L^*_{[\text{RSIR}]} &= \frac{\rho_L (u_L - S_L)}{S_L - S_R} S_M \left(-\frac{S_R - S_M}{S_M - S_L} - 1 \right), \\ (\rho u)_L^*_{[\text{RSIR}]} &= \frac{\rho_L (u_L - S_L)}{S_L - S_R} S_M \left(\frac{S_L - S_R}{S_M - S_L} \right). \end{aligned} \quad (5.20)$$

Finally, the HLLC formulation is recovered,

$$(\rho u)_L^*_{[\text{RSIR}]} = \frac{\rho_L (u_L - S_L)}{S_M - S_L} S_M = (\rho u)_L^*_{[\text{HLLC}]}. \quad (5.21)$$

335 *Energy solution*

The RSIR energy solution reads,

$$(\rho E)_L^*_{[\text{RSIR}]} = (\rho E)_L^*_{[\text{HLL}]} - \omega_R \Psi^{\text{energy}}, \quad (5.22)$$

with

$$(\rho E)_L^*_{[\text{HLL}]} = \frac{[(\rho E + p)u]_R - [(\rho E + p)u]_L + S_L (\rho E)_L - S_R (\rho E)_R}{S_L - S_R}, \quad (5.23)$$

and

$$\Psi^{\text{energy}} = \beta \left(\frac{(\rho E)_R (u_R - S_R) + p_R u_R - p_R^* S_M}{S_M - S_R} - \frac{(\rho E)_L (u_L - S_L) + p_L u_L - p_L^* S_M}{S_M - S_L} \right), \quad (5.24)$$

and

$$\omega_R = \frac{S_R - S_M}{S_R - S_L}. \quad (5.25)$$

340 The combination of these relations leads to,

$$\begin{aligned}
(\rho E)_{L[\text{RSIR}]}^* &= \left[(\rho E + p) u \right]_R \overbrace{\left(\frac{1}{S_L - S_R} - \beta \frac{S_R - S_M}{S_R - S_L} \frac{1}{S_M - S_R} \right)}{=0 \text{ if } \beta=1} \\
&\quad - \left[(\rho E + p) u \right]_L \left(\frac{1}{S_L - S_R} - \beta \frac{S_R - S_M}{S_R - S_L} \frac{1}{S_M - S_L} \right) \\
&\quad + (\rho E)_L \left(\frac{S_L}{S_L - S_R} - \beta \frac{S_R - S_M}{S_R - S_L} \frac{S_L}{S_M - S_L} \right) \\
&\quad - (\rho E)_R \underbrace{\left(\frac{S_R}{S_L - S_R} - \beta \frac{S_R - S_M}{S_R - S_L} \frac{S_R}{S_M - S_L} \right)}_{=0 \text{ if } \beta=1} \\
&\quad - \beta \frac{S_R - S_M}{S_R - S_L} \left(\frac{p_L^* S_M}{S_M - S_L} - \frac{p_R^* S_M}{S_M - S_R} \right).
\end{aligned} \tag{5.26}$$

HLL solution appears recovered when $\beta = 0$. However for $\beta = 1$, the solution becomes,

$$\begin{aligned}
(\rho E)_{L[\text{RSIR}]}^* &= - \left[(\rho E + p) u \right]_L \frac{1}{S_L - S_R} \left(1 + \frac{S_R - S_M}{S_M - S_L} \right) \\
&\quad + (\rho E)_L \frac{S_L}{S_L - S_R} \left(1 + \frac{S_R - S_M}{S_M - S_L} \right) \\
&\quad - \frac{S_R - S_M}{S_R - S_L} \left(\frac{p_L^* S_M}{S_M - S_L} - \frac{p_R^* S_M}{S_M - S_R} \right).
\end{aligned} \tag{5.27}$$

Thus,

$$(\rho E)_{L[\text{RSIR}]}^* = \frac{\left[(\rho E + p) u \right]_L - S_L (\rho E)_L}{S_M - S_L} - \frac{S_M (S_R - S_M)}{S_R - S_L} \left(\frac{p_L^*}{S_M - S_L} - \frac{p_R^*}{S_M - S_R} \right). \tag{5.28}$$

As the Euler equations are considered in this paper, equality of the star pressures is now used, $p_L^* = p_R^* = p^*$. Relation (5.28) results in,

$$(\rho E)_{L[\text{RSIR}]}^* = \frac{\left[(\rho E + p) u \right]_L - S_L (\rho E)_L - p^* S_M}{S_M - S_L} = (\rho E)_{L[\text{HLLC}]}^*, \tag{5.29}$$

345 which is once more the HLLC solution.

Similar manipulations regarding the right solution state vector provide the same conclusion,

$$\mathbf{U}_{R[\text{RSIR}]}^* = \mathbf{U}_{R[\text{HLLC}]}^*. \tag{5.30}$$

It consequently appears that the reconstruction method based on,

$$\begin{cases} \mathbf{U}_{L[\text{RSIR}]}^* = \mathbf{U}_{[\text{HLL}]}^* - \omega_R \boldsymbol{\Psi}, \\ \mathbf{U}_{R[\text{RSIR}]}^* = \mathbf{U}_{[\text{HLL}]}^* + \omega_L \boldsymbol{\Psi}, \end{cases} \quad (5.31)$$

is equivalent to the HLLC solver when the jump vector $\boldsymbol{\Psi}$ is determined through Rankine-Hugoniot relations. The following computed results confirm this observation. Consequently, it appears that the RSIR approach recovers the HLLC solver (or the HLL one as another limit case) and can be degraded at different levels when complex systems of equations are considered, for example when Rankine-Hugoniot relations are not trivial, or contact wave conditions are intricate (this is true for example with two-phase flow equations). In the content of the Euler equations, the RSIR solver based on approximate thermodynamics (Section 3) provides results in close agreement with those of the HLLC solver as shown in the next section.

6. Numerical results

Comparison with the HLLC solver and exact solution is now addressed on various test problems. Let us recall that $\beta = 1$ in all computations, unless stated otherwise. The following test problems are given in Toro (2009) [15] (page 334). In addition, the blast wave test problem of Colella and Woodward (1984) [18] is also addressed (see also [15], page 612), as well as the shock tube test of Leblanc (see Loubère (2005) [19] for example).

In these computations all variables are dimensionless as done in [15], [20]. Note that these test problems are quite severe except tests 6 and 7 that involve a stationary discontinuity and transport of a discontinuity. Those two last tests are nonetheless mandatory to assess methods' accuracy.

All results are provided by the Godunov (1959) [21] first-order scheme,

$$\mathbf{U}_i^{n+1} = \mathbf{U}_i^n - \frac{\Delta t}{\Delta x} \left(\mathbf{F}_{i+\frac{1}{2}}^* - \mathbf{F}_{i-\frac{1}{2}}^* \right), \quad (6.1)$$

where $n + 1$ and n denote two consecutive time steps and superscript $*$ denotes the Riemann problem solution provided by the RSIR and HLLC solvers. Indexes i and $i \pm \frac{1}{2}$ denote respectively the center of the current numerical cell and its corresponding boundaries.

Obviously, higher-order extensions can be considered but add complexity to examine Riemann solvers' accuracy. Note that the Godunov scheme is stable under the conventional CFL condition. For proper assessment, all results are computed using CFL = 0.9 and a coarse mesh made of 100 regular cells. The initial conditions are provided in Table 1.

| Test | x_0 | ρ_L | u_L | p_L | ρ_R | u_R | p_R |
|------|-------|----------|-----------|---------|----------|-----------|---------|
| 1 | 0.3 | 1.0 | 0.75 | 1.0 | 0.125 | 0.0 | 0.1 |
| 2 | 0.5 | 1.0 | -2.0 | 0.4 | 1.0 | 2.0 | 0.4 |
| 3 | 0.5 | 1.0 | 0.0 | 1000.0 | 1.0 | 0.0 | 0.01 |
| 4 | 0.4 | 5.99924 | 19.5975 | 460.894 | 5.99242 | -6.19633 | 46.0950 |
| 5 | 0.8 | 1.0 | -19.59745 | 1000.0 | 1.0 | -19.59745 | 0.01 |
| 6 | 0.5 | 1.4 | 0.0 | 1.0 | 1.0 | 0.0 | 1.0 |
| 7 | 0.5 | 1.4 | 0.1 | 1.0 | 1.0 | 0.1 | 1.0 |

Table 1: Initial conditions of the 7 test problems of Toro (2009) [15] page 334. The computational domain is the interval $[0, 1]$ for all tests. An initial discontinuity is located at x_0 . All initial conditions are dimensionless. The initial conditions of Colella and Woodward’s (1984) [18] test problem are given in Fig. 9 (see also [15], page 612). The initial conditions of Leblanc’s test are given in Figs. 10 and 11.

375 In the following figures, the RSIR solver based on approximate thermodynamics (Section 3) is compared to the exact and HLLC solutions. In addition, the RSIR solution based on Rankine-Hugoniot relations is plotted as well and is referred as “RSIR RH”. The solution of the original RSIR solver introduced in Carmouze et al. (2020) [8] is also plotted and is referred as “RSIR orig.”. Note that the exact solution is computed with an exact solver and then plotted with 1000 points.

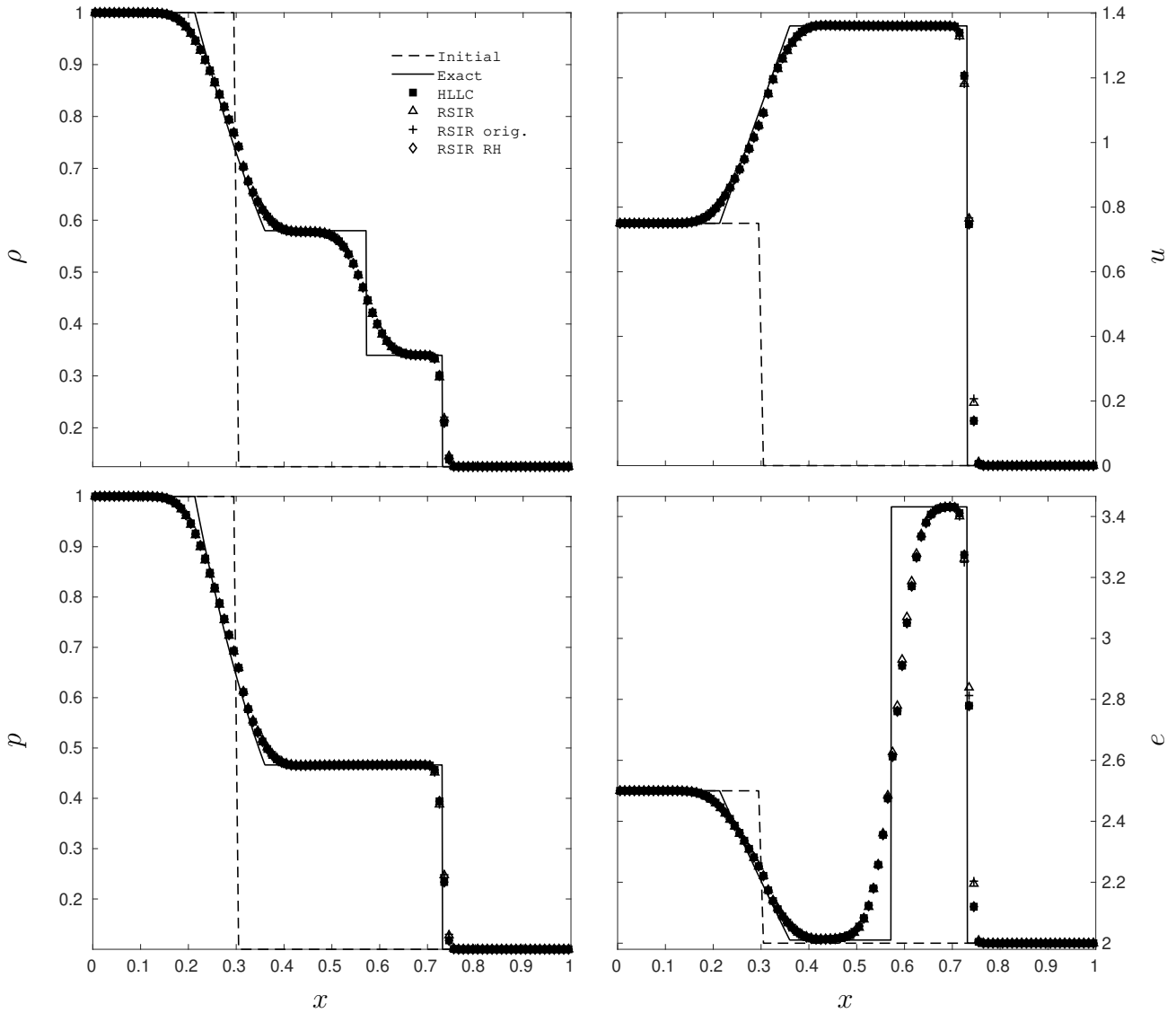


Figure 2: Test 1 of [15] page 334 (shock tube test). Comparison of the RSIR, HLLC and exact solutions. The computational domain involves 100 cells and the Godunov first-order scheme is used with CFL = 0.9. Results are shown at time $t = 0.2$. The initial discontinuity is located at $x_0 = 0.3$. Non-reflective boundary conditions are considered. All methods show comparable accuracy.

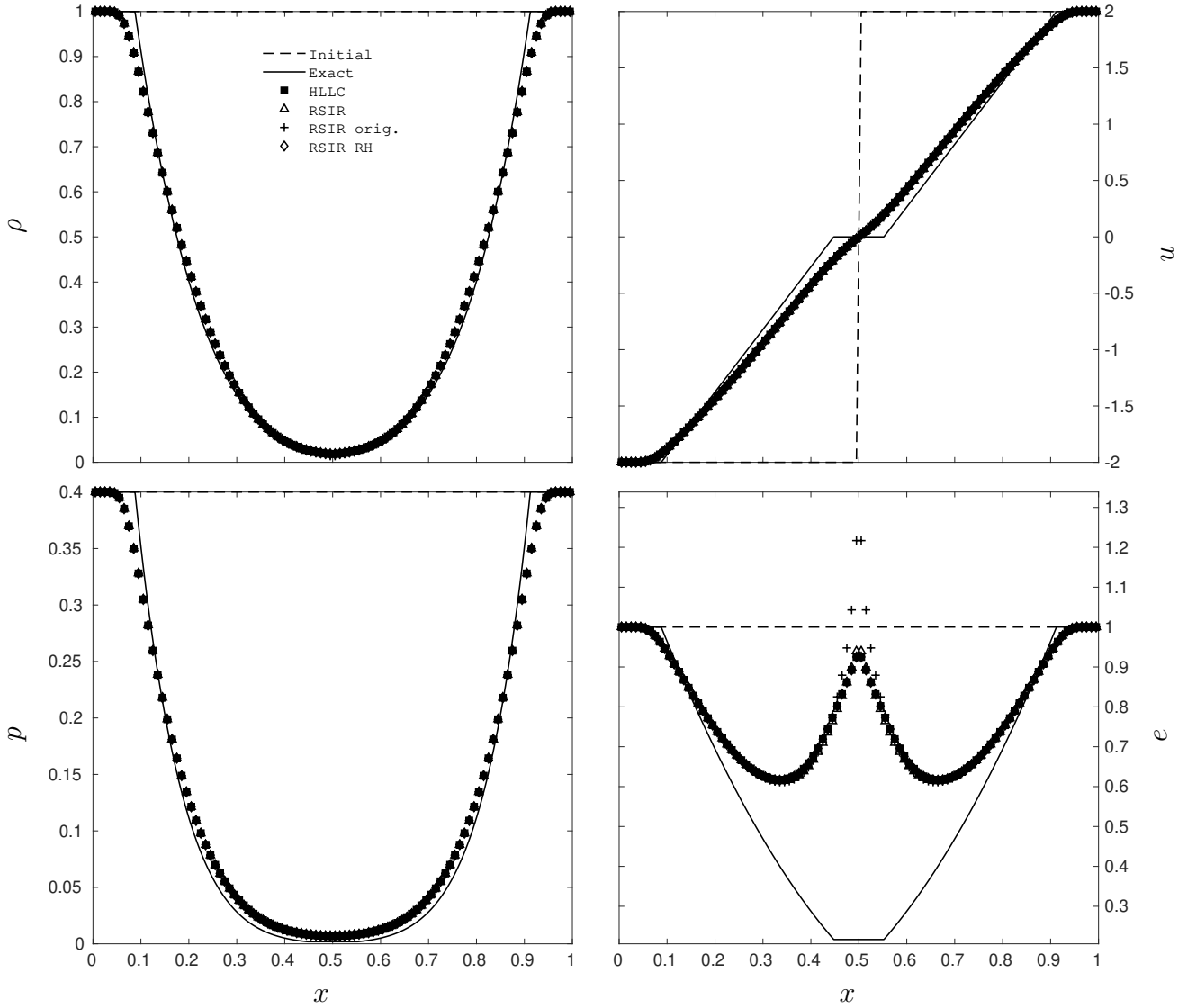


Figure 3: Test 2 of [15] page 334 (double expansion test). Comparison of the RSIR, HLLC and exact solutions. The computational domain involves 100 cells and the Godunov first-order scheme is used with $CFL = 0.9$. Results are shown at time $t = 0.15$. The initial discontinuity is located at $x_0 = 0.5$. Non-reflective boundary conditions are considered. All methods produce unphysical overheating at the center of the domain regarding the internal energy. The other flow variables are computed correctly. Issues related to overheating do not seem related to the Riemann solver's accuracy but more to the computation of the kinetic energy (Cocchi et al. (1998) [22]). Improvement of the present version of the RSIR solver is clearly visible in the internal energy plot, compared to the original version of Carmouze et al. (2020) [8].

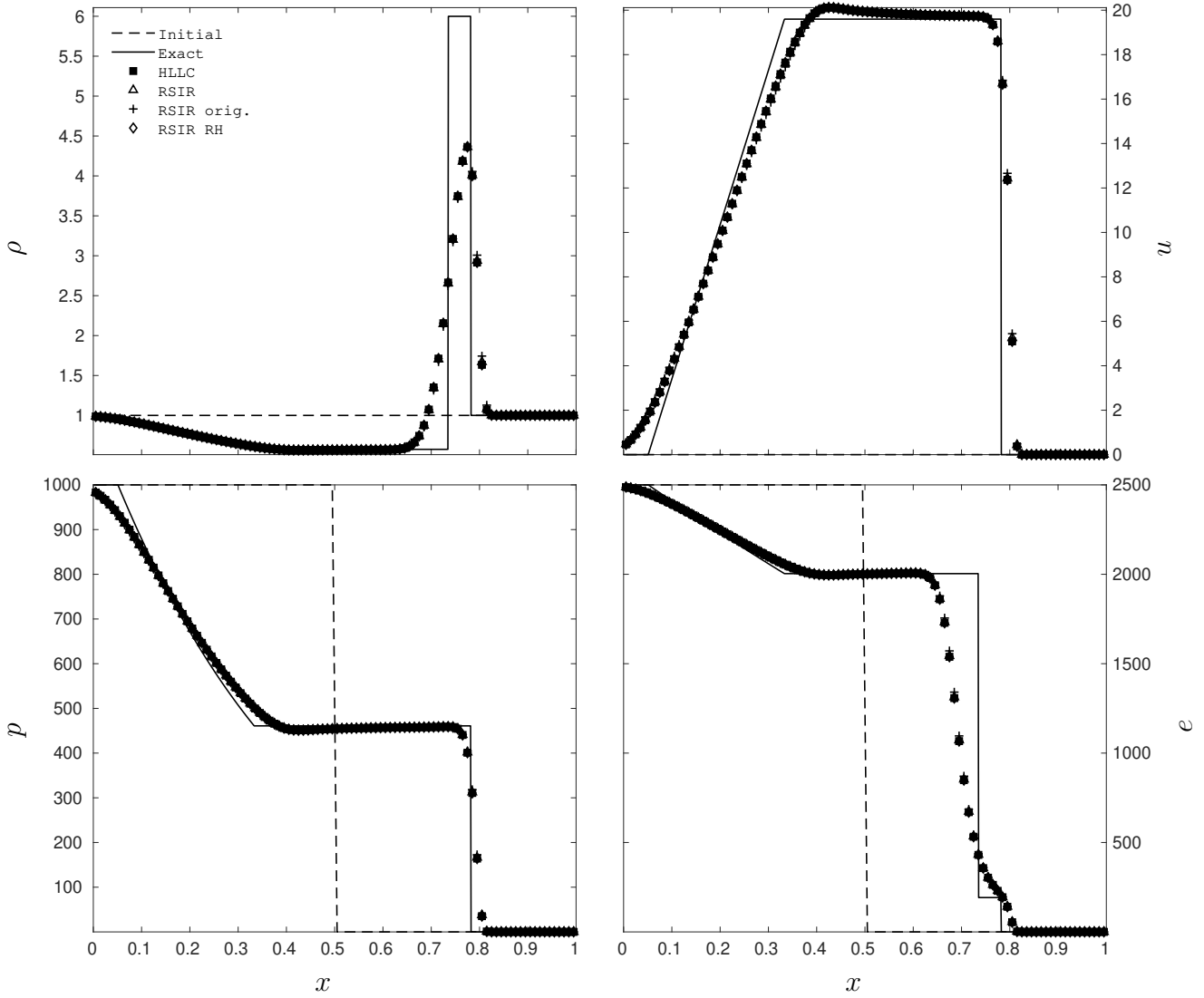


Figure 4: Test 3 of [15] page 334 (strong shock tube test). Comparison of the RSIR, HLLC and exact solutions. The computational domain involves 100 cells and the Godunov first-order scheme is used with $CFL = 0.9$. Results are shown at time $t = 0.012$. The initial discontinuity is located at $x_0 = 0.5$. Non-reflective boundary conditions are considered. All methods show comparable accuracy.

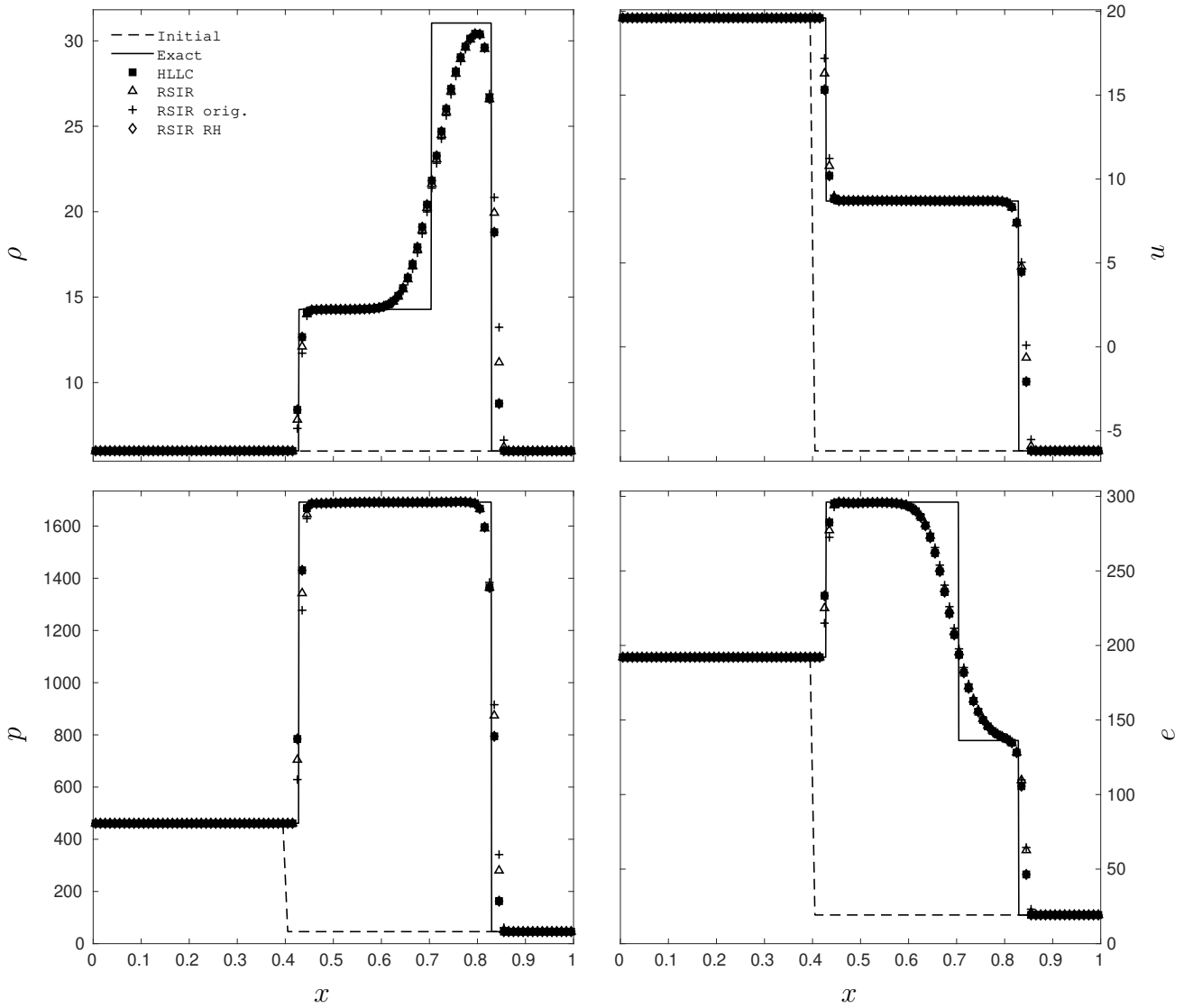


Figure 5: Test 4 of [15] page 334 (double shock test). Comparison of the RSIR, HLLC and exact solutions. The computational domain involves 100 cells and the Godunov first-order scheme is used with $CFL = 0.9$. Results are shown at time $t = 0.035$. The initial discontinuity is located at $x_0 = 0.4$. Non-reflective boundary conditions are considered. All methods show comparable accuracy.

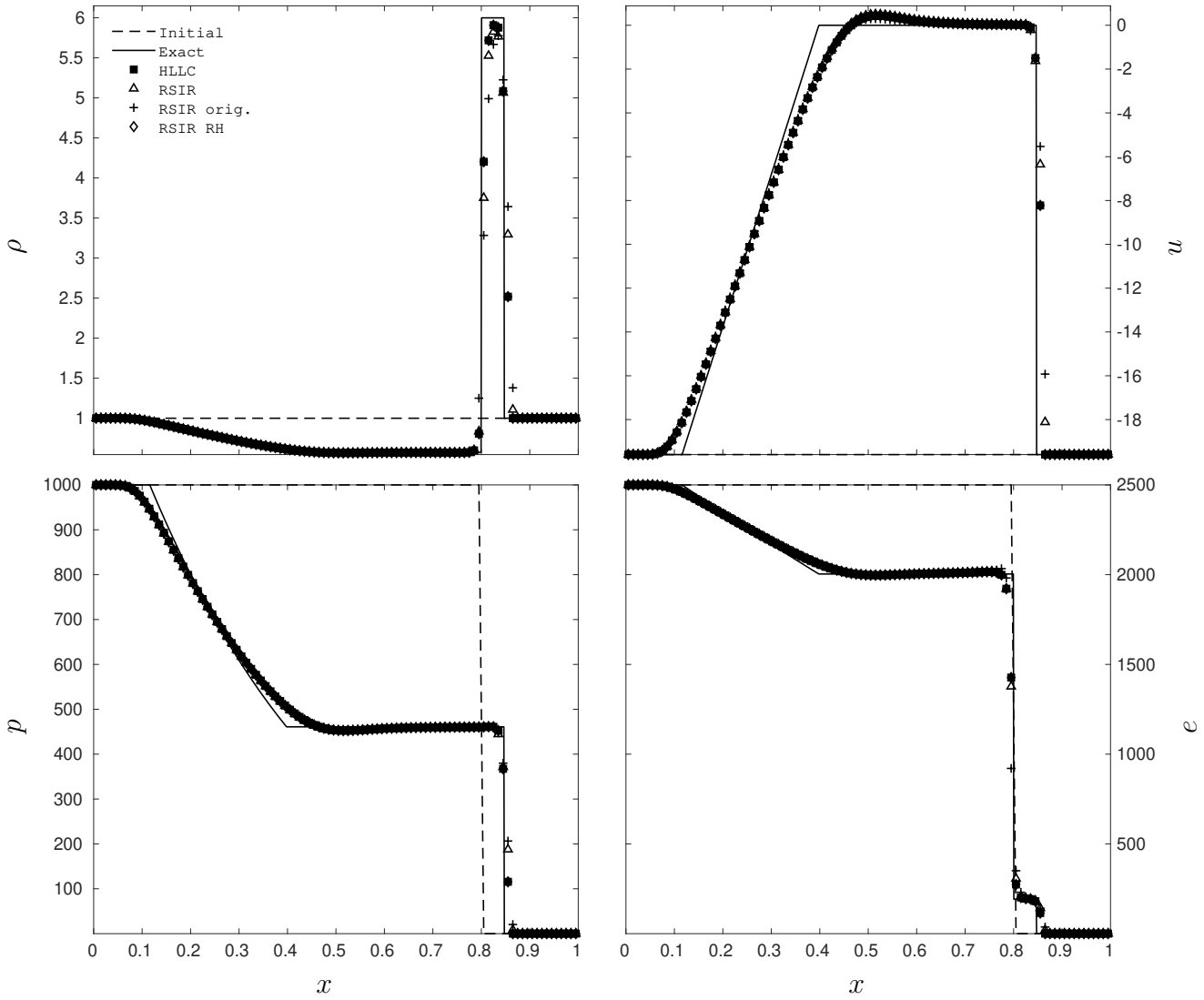


Figure 6: Test 5 of [15] page 334 (strong shock tube test of Test 3 with non-zero initial velocity). Comparison of the RSIR, HLLC and exact solutions. The computational domain involves 100 cells and the Godunov first-order scheme is used with $CFL = 0.9$. Results are shown at time $t = 0.012$. The initial discontinuity is located at $x_0 = 0.8$. Non-reflective boundary conditions are considered. All methods show comparable accuracy.

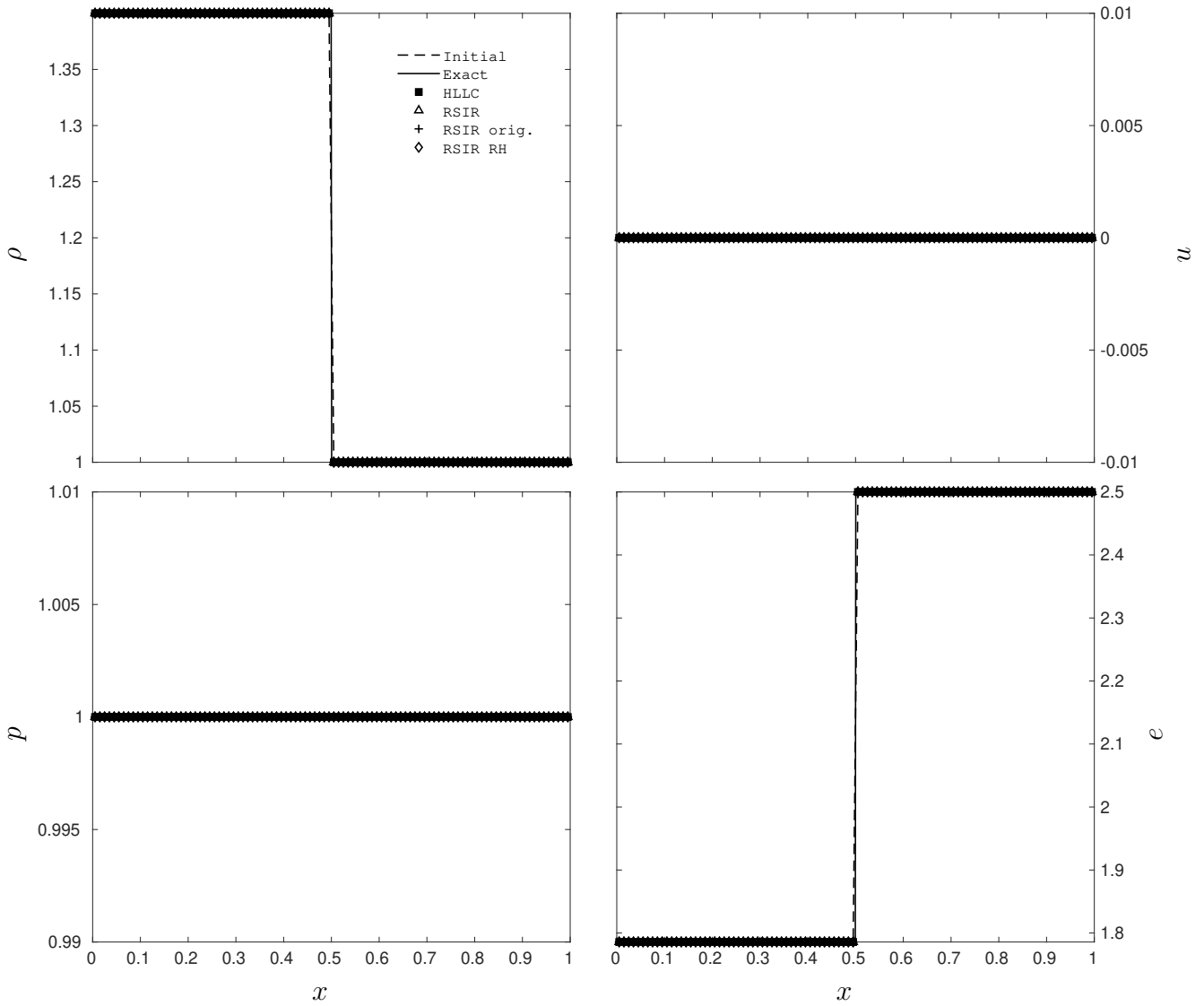


Figure 7: Test 6 of [15] page 334 (stationary contact discontinuity). Comparison of the RSIR, HLLC and exact solutions. The computational domain involves 100 cells and the Godunov first-order scheme is used with $CFL = 0.9$. Results are shown at time $t = 2$. The initial discontinuity is located at $x_0 = 0.5$. Non-reflective boundary conditions are considered. All methods maintain stationary contact discontinuity.

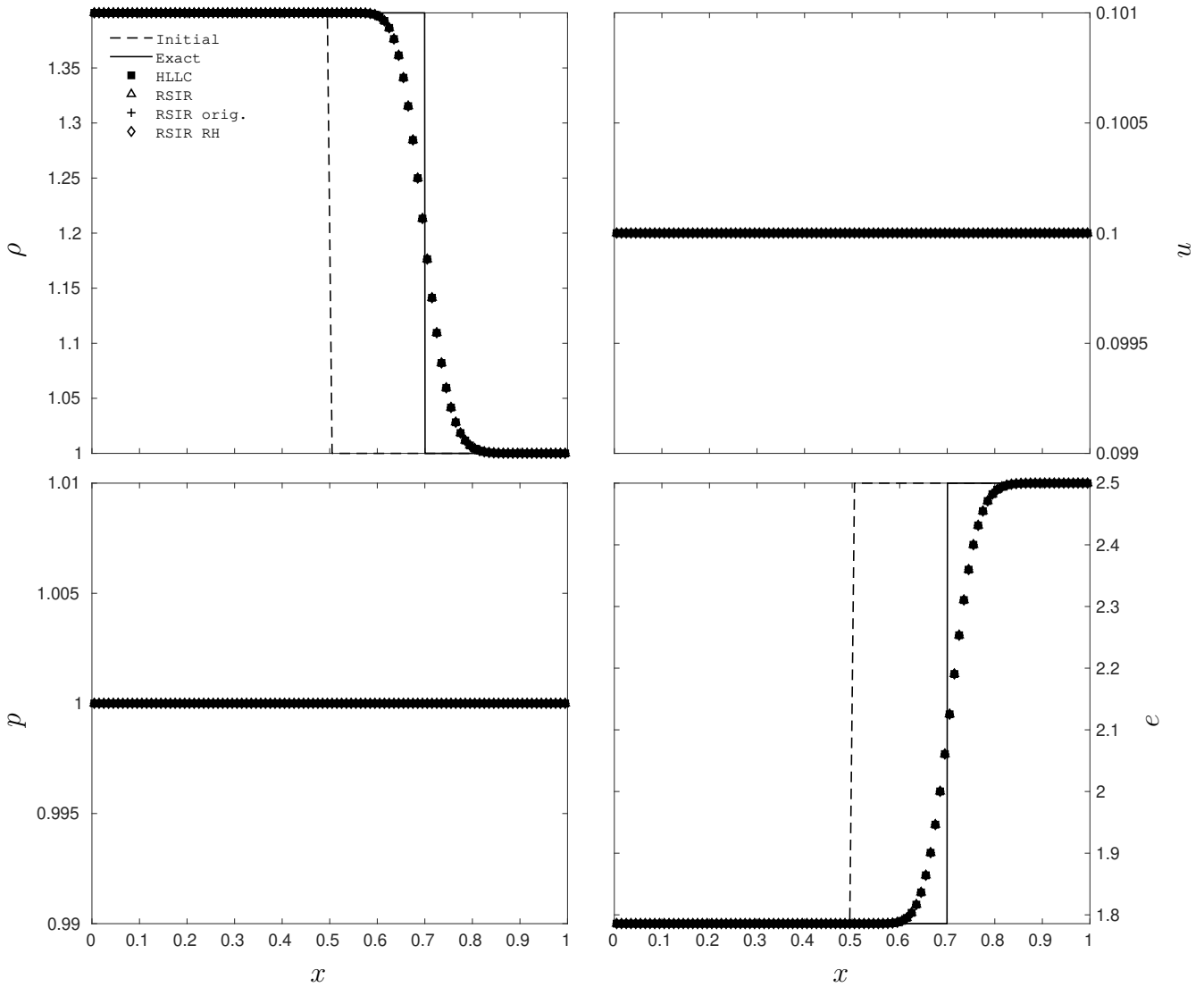


Figure 8: Test 7 of [15] page 334 (moving contact discontinuity). Comparison of the RSIR, HLLC and exact solutions. The computational domain involves 100 cells and the Godunov first-order scheme is used with $CFL = 0.9$. Results are shown at time $t = 2$. The initial discontinuity is located at $x_0 = 0.5$. Non-reflective boundary conditions are considered. All methods show comparable accuracy.

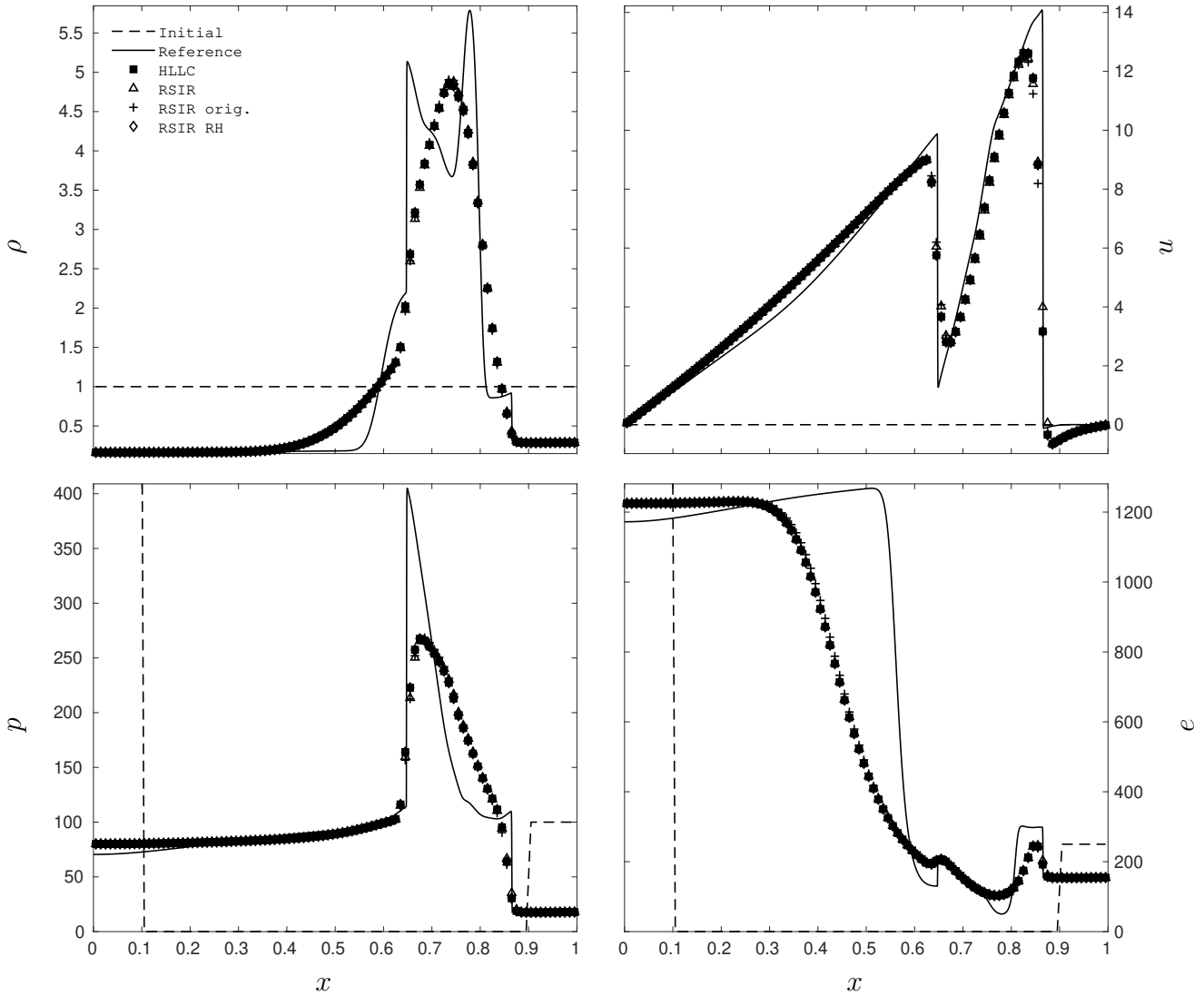


Figure 9: Woodward and Colella's (1984) [18] blast wave test (see also Toro (2009) [15] page 612). Comparison of the RSIR and HLLC solutions. The computational domain involves 100 cells and the Godunov first-order scheme is used with $CFL = 0.9$. Results are shown at time $t = 0.038$. This test problem has no exact solution, but computed results with the HLLC solver and 3000 cells are shown as reference solution. The computational domain is the interval $[0, 1]$. The initial condition in pressure consists of three constant states separated by two discontinuities, namely $p(x, 0) = p_L = 1000$ for $x < 1/10$, $p(x, 0) = p_M = 0.01$ for $1/10 < x < 9/10$ and $p(x, 0) = p_R = 100$ for $x > 9/10$. Particle velocity and density are constant, with $u(x, 0) = 0$ and $\rho(x, 0) = 1$. Reflective boundary conditions are imposed. Note that for the sake of clarity the scale is adapted. All methods show comparable accuracy.

380 All tests lead to the same observations. The RSIR solver provides excellent and oscillation-free results in addition to being robust and positive. Its limit based on Rankine-Hugoniot relations provides results in perfect agreement with the HLLC solver. This observation confirms the analysis of Section 5.

385 The present RSIR solver is an improvement of the original version introduced in Carmouze et al. (2020) [8] as clearly seen in Fig. 3 where the unphysical overheating of the internal energy is significantly reduced, providing results in close agreement with those of the HLLC solver. This improvement is due to the introduction of Gibbs' identity in the form (3.18) when computing the energy jump.

390 Unlike the HLLC solver, the RSIR one is based on thermodynamics and internal reconstruction. The discrete entropy production is satisfied and is quantified by the average sound speed \bar{c}^2 . In all previous tests, $\bar{c}^2 = \max [c_L^2, c_R^2]$ in order to maximize the discrete entropy production for the sake of robustness. This flexibility is an interesting feature. For instance tests 3, 4, 5 and 8 fail if $\bar{c}^2 = \min [c_L^2, c_R^2]$ (Figs. 4, 5, 6 and 9) while tests 1, 2, 6 and 7 provide similar results (Figs. 2, 3, 7 and 8).

395 Flexibility and simplicity of the RSIR solver are of particular interest for complicated flow models. In the context of the Euler equations, the RSIR solver recovers the HLL solver in some limit and the HLLC one in another limit.

The shock tube test of Leblanc is now addressed. Due to the extreme conditions of this test, parameter β had to be set to 0 during the first time step. The rest of the computation is done 400 with $\beta = 1$. Results are provided in Fig. 10.

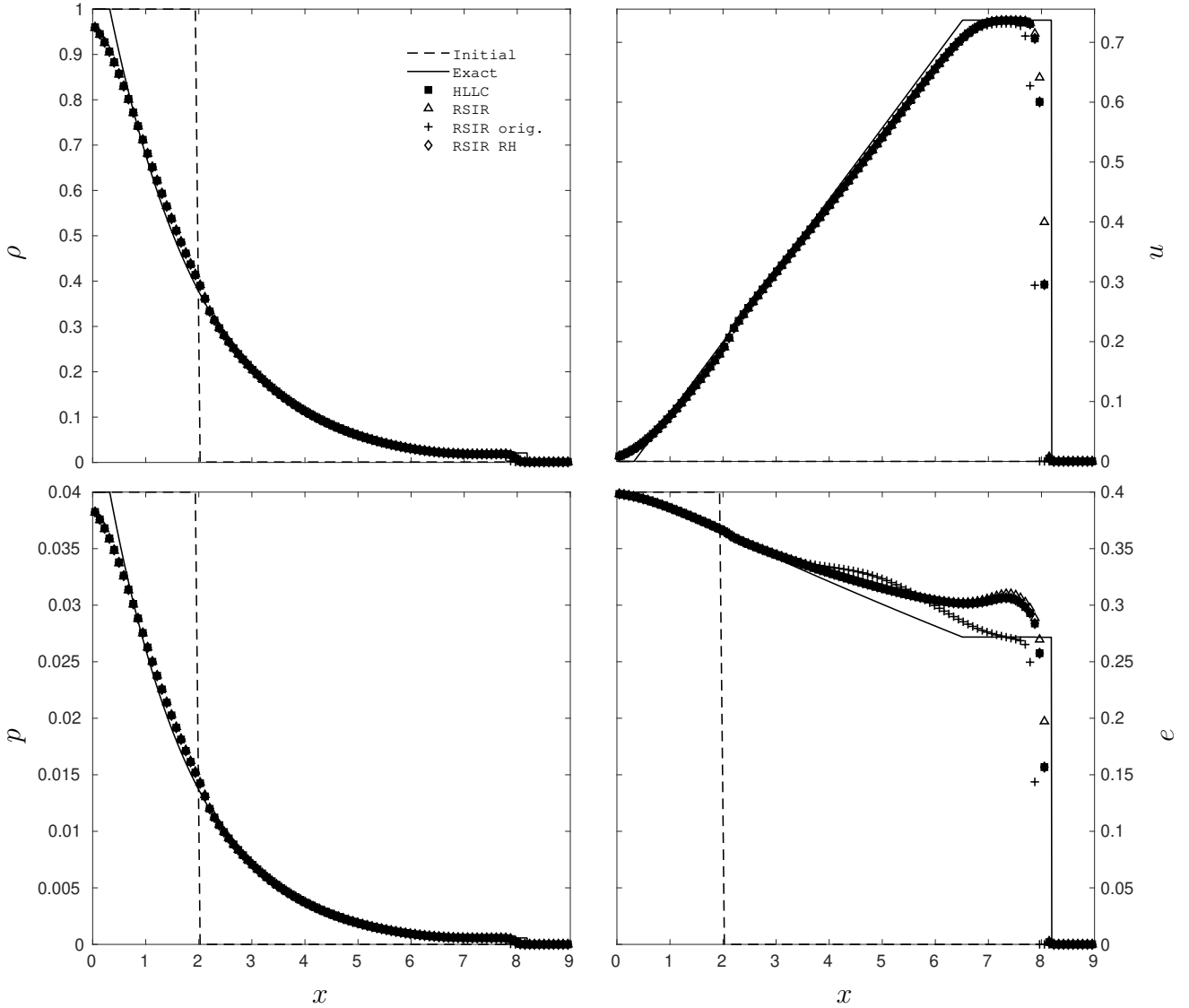


Figure 10: Leblanc’s shock tube test. Comparison of the RSIR and HLLC solutions. The computational domain involves 100 cells and the Godunov first-order scheme is used with $CFL = 0.9$. Results are shown at time $t = 8$. The computational domain is the interval $[0, 9]$. The initial discontinuity is located at $x_0 = 2$. Non-reflective boundary conditions are considered. On the left of the discontinuity, the initial conditions are $\rho_L = 1$, $p_L = 4 \cdot 10^{-2}$, $u_L = 0$. On the right of the discontinuity, the initial conditions are $\rho_R = 10^{-3}$, $p_R = 4 \cdot 10^{-11}$, $u_R = 0$. The ideal gas equation of state is used with $\gamma = 1.1$. Davis’ estimates are used for all computations. However, the severity of this test demands to set $\beta = 0$ for the RSIR solver (triangle symbols Δ) for the first time step only. The rest of the computation is performed with $\beta = 1$.

The lack of robustness at the first time step can be corrected with better extreme wave speed estimates. In the present work Davis’ estimates (1988) [16] are used, Eq. (2.2). Those are simple and convenient for complex flow models as well as sophisticated equations of state. They however seem insufficiently accurate for Leblanc’s test problem. The estimates of Toro (2009)

405 [15] are more accurate but require the equation of state. Toro's estimates read:

$$\begin{cases} S_L = u_L - c_L q_L, \\ S_R = u_R + c_R q_R, \end{cases} \quad (6.2)$$

with,

$$q_k = \begin{cases} 1 & \text{if } p^* \leq p_k, \\ \left(1 + \frac{\gamma + 1}{2\gamma} \left(\frac{p^*}{p_k}\right) - 1\right)^{\frac{1}{2}} & \text{if } p^* > p_k, \end{cases} \quad (6.3)$$

and,

$$\begin{cases} p^* = \max[0, p_{vrs}], & p_{vrs} = \frac{1}{2}(p_L + p_R) - \frac{1}{2}(u_L - u_R)\bar{\rho}\bar{c}, \\ \bar{\rho} = \frac{1}{2}(\rho_L + \rho_R), & \bar{c} = \frac{1}{2}(c_L + c_R). \end{cases} \quad (6.4)$$

These estimates discriminate shocks and rarefactions. If the wave is a rarefaction, the speed is the fastest signal, *i.e.* the head of the rarefaction. If the wave is a shock, then the speed
 410 approximates the shock speed. Results are provided in Fig. 11.

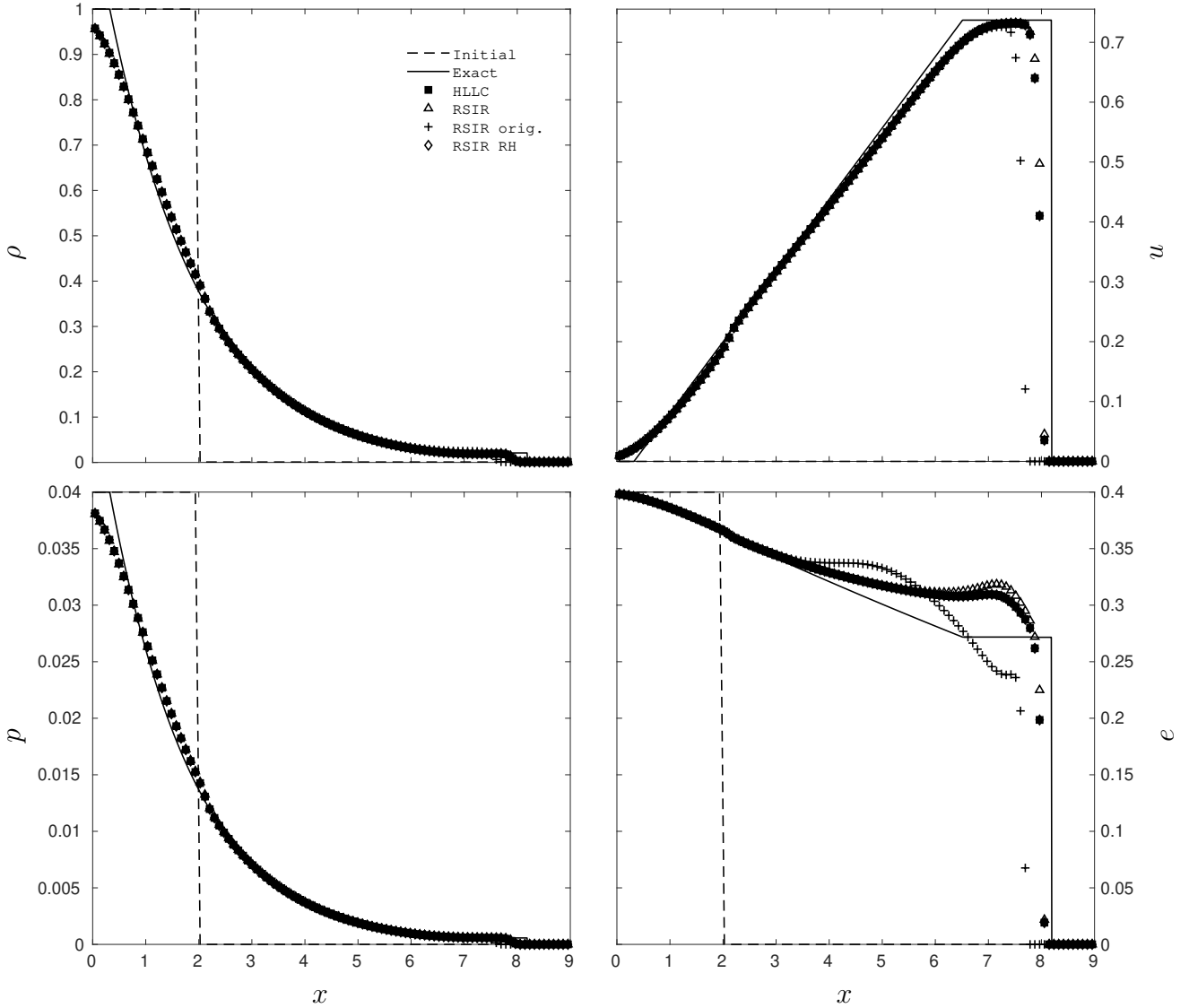


Figure 11: Leblanc’s shock tube test. Comparison of the RSIR and HLLC solutions. The computational domain involves 100 cells and the Godunov first-order scheme is used with $CFL = 0.9$. Results are shown at time $t = 8$. The computational domain is the interval $[0, 9]$. The initial discontinuity is located at $x_0 = 2$. Non-reflective boundary conditions are considered. On the left of the discontinuity, the initial conditions are $\rho_L = 1$, $p_L = 4 \cdot 10^{-2}$, $u_L = 0$. On the right of the discontinuity, the initial conditions are $\rho_R = 10^{-3}$, $p_R = 4 \cdot 10^{-11}$, $u_R = 0$. The ideal gas equation of state is used with $\gamma = 1.1$. Toro’s estimates are used for all computations. Thanks to these more accurate estimates, the RSIR solver uses $\beta = 1$ for the whole computation.

Two-dimensional illustration

Two-dimensional computations are addressed with the following example of shock wave interaction with an inclined plate. The test problem is described in Toro’s textbook [15], page 593. The full domain is 25 m long and 16.5 m height. It is meshed with 442,793 unstructured

415 triangles. The inclined plate begins at abscissa 4.69 m with angle 25 degrees. A plane shock wave is initially settled at $x = 4$ m and propagates at Mach number 1.7. Computed results are shown at time 0.02 s with 100 density contours. MUSCL reconstruction is used with van Leer's limiter [23]. Results are shown in Fig. 12.

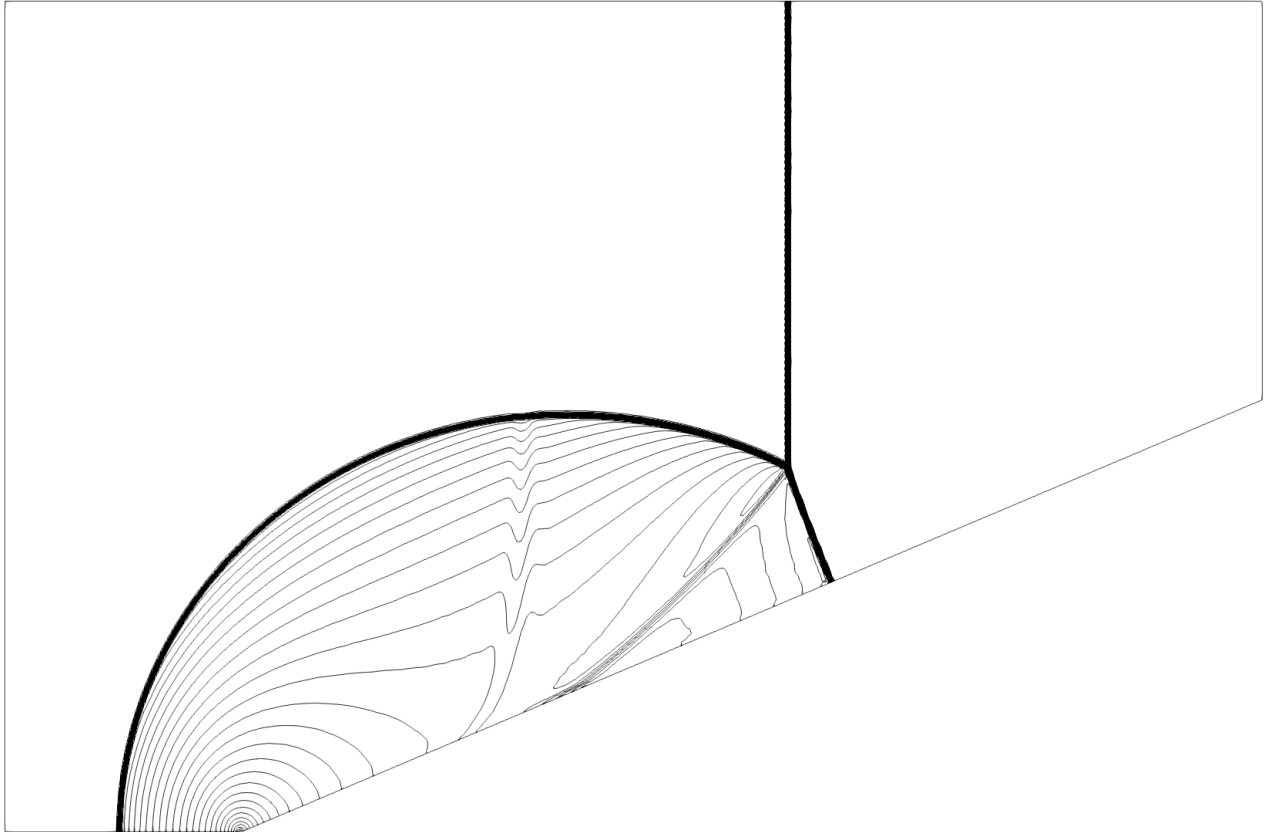


Figure 12: Shock wave diffraction problem for $M = 1.7$ and wedge angle of 25 degrees. The RSIR solver is used. The solution is given at time 0.02 s with 100 density contours. An unstructured mesh made of 442,793 triangles is used. MUSCL reconstruction is used with van Leer's limiter [23]. Reflective boundary conditions are considered except for the left one that is non-reflective.

420 The wave pattern of the solution corresponds to single Mach reflection. There are three shocks meeting at the triple point, namely, part of the incident shock, the reflected shock and the Mach stem, which joins the triple point with the wedge surface. From the triple point there emerges a slip surface that joins the wedge surface at a sharp angle. The present computational results qualitatively agree with experimental results, see Toro's textbook [15], page 594.

7. Conclusion

425 The Riemann solver with internal reconstruction (RSIR) of Carmouze et al. (2020) [8] has been investigated, revisited and improved for the Euler equations. While the RSIR solver based

on approximate thermodynamics produces accurate and robust results, similar to those provided by the HLLC solver of Toro et al. (1994) [5], its limit using the whole set of Rankine-Hugoniot relations appears strictly equivalent to the HLLC solver.

430 RSIR-type solvers seem flexible for many applications where most of the physics is governed by the two extreme waves and an intermediate one, as they can be simplified at different levels when complex systems of equations are addressed. Two-phase flow models are for instance part of those applications as seen in Carmouze et al. (2020) [8] and Chiapolino and Saurel (2020) [14].

435 Other two-phase flow models can be addressed as well, such as Kapila et al.’s (2001) [24] model with the two-pressure formulation given in Saurel et al. (2009) [25]. Models in mechanical and thermal equilibrium, such as the one given in Saurel et al. (2016) [26] can be addressed as well.

Regarding the Discrete Equations Method (DEM) (Abgrall and Saurel (2003) [27]), widely used to address Baer and Nunziato (1986) [28] (BN) type models, the extension seems non-trivial. Indeed, interfaces separating fluids having very large differences in densities, sound speeds and acoustic impedances may be present at internal boundaries. There is no evidence that the average sound speed used in RSIR would be appropriate. However, RSIR extension seems possible for the BN-type model obtained in the continuous limit of the DEM (Saurel et al. (2003) [29]), following the same lines as Carmouze et al. (2020) [8].

445 Regarding low Mach number computations, where $\bar{c} \rightarrow +\infty$ specific analysis is needed. In such situation, right- and left-facing wave speeds must be modified (Guillard and Viozat, (1999) [30]) with a sound speed function of the local Mach number. There is no evidence that the same sound speed can be used in \bar{c} . Deeper investigations are needed.

450 Regarding hyper-elastic flow models, such as Gavrilyuk et al.’s (2008) [2], the extension of RSIR seems quite easy. Regarding MHD models (Balsara et al. (2012) [1]) the authors are not specialist enough to provide recommendations.

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They are also very grateful to the anonymous referees for the quality of their reports that definitely helped to improve the quality of this contribution.

Appendix A. Summary of the RSIR solver

460 The present appendix summarizes the RSIR formulas for a straightforward implementation. The two extreme wave speeds are first computed. Davis’ estimates are used in the present contribution,

$$S_L = \min(u_L - c_L, u_R - c_R) \quad \text{and} \quad S_R = \max(u_L + c_L, u_R + c_R). \quad (\text{A.1})$$

The HLL solution state is then computed,

$$\mathbf{U}_{[\text{HLL}]}^* = \frac{\mathbf{F}_R - \mathbf{F}_L + S_L \mathbf{U}_L - S_R \mathbf{U}_R}{S_L - S_R}. \quad (\text{A.2})$$

The contact wave speed S_M is now required. In the context of the Euler equations, it reads,

$$S_M = u_L^* = u_R^* = \frac{p_R - p_L + (\rho u)_L (S_L - u_L) - (\rho u)_R (S_R - u_R)}{\rho_L (S_L - u_L) - \rho_R (S_R - u_R)} = \frac{U_{[\text{HLL}]}^{*,\text{momentum}}}{U_{[\text{HLL}]}^{*,\text{mass}}}. \quad (\text{A.3})$$

The two “weights” ω_L and ω_R are then computed,

$$\omega_R = \frac{S_R - S_M}{S_R - S_L} \quad \text{and} \quad \omega_L = \frac{S_M - S_L}{S_R - S_L}. \quad (\text{A.4})$$

465 The solution densities are now determined with the help of the mass jump and the consistency relation,

$$\begin{cases} \rho_L^* = \rho_{[\text{HLL}]}^* - \omega_R \beta \left(\rho_R - \rho_L + \frac{p_L - p_R}{\bar{c}^2} \right), \\ \rho_R^* = \rho_{[\text{HLL}]}^* + \underbrace{\omega_L \beta \left(\rho_R - \rho_L + \frac{p_L - p_R}{\bar{c}^2} \right)}_{\Psi^{\text{mass}}}, \end{cases} \quad (\text{A.5})$$

where the average sound speed reads,

$$\bar{c}^2 = \max [c_L^2, c_R^2]. \quad (\text{A.6})$$

The solution momenta are similarly obtained,

$$\begin{cases} (\rho u)_L^* = (\rho u)_{[\text{HLL}]}^* - \omega_R \beta \left(\rho_R - \rho_L + \frac{p_L - p_R}{\bar{c}^2} \right) S_M, \\ (\rho u)_R^* = (\rho u)_{[\text{HLL}]}^* + \underbrace{\omega_L \beta \left(\rho_R - \rho_L + \frac{p_L - p_R}{\bar{c}^2} \right) S_M}_{\Psi^{\text{momentum}}}. \end{cases} \quad (\text{A.7})$$

The total energy solutions are then computed,

$$\begin{cases} (\rho E)_L^* = (\rho E)_{[\text{HLL}]}^* - \omega_R \beta \left[\rho_R^* \left(e_R^* + \frac{1}{2} S_M^2 \right) - \rho_L^* \left(e_L^* + \frac{1}{2} S_M^2 \right) \right], \\ (\rho E)_R^* = (\rho E)_{[\text{HLL}]}^* + \underbrace{\omega_L \beta \left[\rho_R^* \left(e_R^* + \frac{1}{2} S_M^2 \right) - \rho_L^* \left(e_L^* + \frac{1}{2} S_M^2 \right) \right]}_{\Psi^{\text{energy}}}, \end{cases} \quad (\text{A.8})$$

470 where ρ_R^* and ρ_L^* are given by Eq. (A.5) and e_R^* and e_L^* are provided by,

$$\begin{cases} e_L^* = e_L - p^* (v_L^* - v_L), \\ e_R^* = e_R - p^* (v_R^* - v_R), \end{cases} \quad (\text{A.9})$$

where $v_L^* = 1/\rho_L^*$ and $v_R^* = 1/\rho_R^*$ and,

$$p^* = p_L + \bar{c}^2 (\rho_L^* - \rho_L) = p_R + \bar{c}^2 (\rho_R^* - \rho_R). \quad (\text{A.10})$$

The intermediate solution states are then fully determined for the Euler equations. They are based upon the HLL solution with the help of the consistency relation and the jump vector,

$$\begin{cases} \mathbf{U}_{L[\text{RSIR}]}^* = \mathbf{U}_{[\text{HLL}]}^* - \omega_R \Psi, \\ \mathbf{U}_{R[\text{RSIR}]}^* = \mathbf{U}_{[\text{HLL}]}^* + \omega_L \Psi. \end{cases} \quad (\text{A.11})$$

475 The various fluxes are finally computed through the Rankine-Hugoniot relations, according to the sign of S_M ,

$$\begin{cases} \mathbf{F}_R^* = \mathbf{F}_R + S_R (\mathbf{U}_R^* - \mathbf{U}_R), \\ \mathbf{F}_L^* = \mathbf{F}_L + S_L (\mathbf{U}_L^* - \mathbf{U}_L). \end{cases} \quad (\text{A.12})$$

The flux solution is then sampled according to the procedure,

$$\mathbf{F}^* = \begin{cases} \mathbf{F}_L & \text{if } S_L > 0, \\ \mathbf{F}_L^* & \text{if } S_L < 0 \quad \text{and } S_M > 0, \\ \mathbf{F}_R^* & \text{if } S_M < 0 \quad \text{and } S_R > 0, \\ \mathbf{F}_R & \text{otherwise.} \end{cases} \quad (\text{A.13})$$

AppendixB. β as a viscosity parameter

The combination of Eqs. (A.2), (A.4) and (A.5) results for the mass relation in,

$$\rho_L^* = \frac{(\rho u)_R - (\rho u)_L + S_L \rho_L - S_R \rho_R}{S_L - S_R} + \left(\frac{S_R - S_M}{S_L - S_R} \right) \beta \left(\rho_R - \rho_L + \frac{p_L - p_R}{\bar{c}^2} \right). \quad (\text{B.1})$$

480 For the sake of simplicity, let us consider a stationary discontinuity in atmospheric conditions. In that context, the speeds are zero: $u_L = u_R = S_M = 0$ and pressures are uniform $p_L = p_R = p^*$. The preceding relation consequently simplifies to,

$$\rho_L^* = \frac{\rho_L (S_L - \beta S_R) + \rho_R (S_R \beta - S_R)}{S_L - S_R}. \quad (\text{B.2})$$

The solution pressure of the RSIR solver based on approximate thermodynamics reads,

$$p^* = p_L + \bar{c}^2 (\rho_L^* - \rho_L) = p_R + \bar{c}^2 (\rho_R^* - \rho_R). \quad (\text{B.3})$$

In order to maintain uniform pressure $p^* = p_L$, then necessary $\rho_L^* = \rho_L$. Equation (B.2) satisfies this condition only for $\beta = 1$. The same reasoning with the right solution state ρ_R^* yields the same conclusion.

When $\beta = 0$, the HLL solution is recovered as clearly seen with Eq. (A.5). It thus appears that the only admissible values are $\beta = 0$ and $\beta = 1$ for the RSIR solver based on thermodynamics. However, this restriction does not apply to the variant version RSIR RH. Indeed, this variant is based on the Rankine-Hugoniot relations and the pressure solution reads,

$$p_k^* = p_k + \rho_k (S_k - u_k) (S_M - u_k), \quad k = L, R. \quad (\text{B.4})$$

Parameter β has no involvement in Eq. (B.4). When $u_L = u_R = S_M = 0$, the pressure remains uniform $p_L = p_R = p^*$. Variant RSIR RH consequently admits $0 \leq \beta \leq 1$, the two limits corresponding to the HLL and HLLC solutions respectively. Numerical viscosity induced by parameter β is now addressed.

The Godunov scheme reads,

$$\mathbf{U}_i^{n+1} = \mathbf{U}_i^n - \frac{\Delta t}{\Delta x} \left(\mathbf{F}_{i+\frac{1}{2}}^* - \mathbf{F}_{i-\frac{1}{2}}^* \right). \quad (\text{B.5})$$

The flux, solution of the Riemann problem, is sampled as:

$$\mathbf{F}^* = \begin{cases} \mathbf{F}_L & \text{if } S_L > 0, \\ \mathbf{F}_L^* & \text{if } S_L < 0 \quad \text{and } S_M > 0, \\ \mathbf{F}_R^* & \text{if } S_M < 0 \quad \text{and } S_R > 0, \\ \mathbf{F}_R & \text{otherwise,} \end{cases} \quad (\text{B.6})$$

with

$$\begin{cases} \mathbf{F}_R^* = \mathbf{F}_R + S_R (\mathbf{U}_R^* - \mathbf{U}_R), \\ \mathbf{F}_L^* = \mathbf{F}_L + S_L (\mathbf{U}_L^* - \mathbf{U}_L). \end{cases} \quad (\text{B.7})$$

For the sake of simplicity, S_L , S_R and S_M are assumed uniform, *i.e.* constant everywhere. Moreover, it is assumed: $S_L = -S < 0$, $S_R = S > 0$ and $S_M = u > 0$. As the velocity is uniform, the pressure is uniform as well. Only density discontinuities are assumed present.

With these simplifications the Godunov method reads,

$$\mathbf{U}_i^{n+1} = \mathbf{U}_i^n - \frac{\Delta t}{\Delta x} \left(\mathbf{F}_i - S (\mathbf{U}_{i,L}^* - \mathbf{U}_i) - \mathbf{F}_{i-1} + S (\mathbf{U}_{i-1,L}^* - \mathbf{U}_{i-1}) \right). \quad (\text{B.8})$$

Thus,

$$\mathbf{U}_i^{n+1} = \mathbf{U}_i^n - \frac{\Delta t}{\Delta x} (\mathbf{F}_i - \mathbf{F}_{i-1}) + \frac{\Delta t S}{\Delta x} \left((\mathbf{U}_{i,L}^* - \mathbf{U}_i) - (\mathbf{U}_{i-1,L}^* - \mathbf{U}_{i-1}) \right). \quad (\text{B.9})$$

Let us now examine the star states:

$$\mathbf{U}_{i,L}^* [\text{RSIR}] = \mathbf{U}_i^* [\text{HLL}] - \omega_R \Psi_i, \quad (\text{B.10})$$

with,

$$\mathbf{U}_i^*_{[\text{HLL}]} = \frac{\mathbf{F}_{i+1} - \mathbf{F}_i + S_L \mathbf{U}_i - S_R \mathbf{U}_{i+1}}{S_L - S_R} = \frac{1}{2} (\mathbf{U}_i + \mathbf{U}_{i+1}) - \frac{1}{2S} (\mathbf{F}_{i+1} + \mathbf{F}_i). \quad (\text{B.11})$$

The difference $\mathbf{U}_i^* - \mathbf{U}_i$ reads,

$$\mathbf{U}_i^*_{[\text{RSIR}]} - \mathbf{U}_i = \frac{1}{2} (\mathbf{U}_{i+1} - \mathbf{U}_i) - \frac{1}{2S} (\mathbf{F}_{i+1} - \mathbf{F}_i) - \omega_R \Psi_i. \quad (\text{B.12})$$

505 The Godunov scheme becomes,

$$\mathbf{U}_i^{n+1} = \mathbf{U}_i^n - \frac{\Delta t}{\Delta x} (\mathbf{F}_i - \mathbf{F}_{i-1}) + \frac{\Delta t S}{2\Delta x} \left((\mathbf{U}_{i+1} - 2\mathbf{U}_i + \mathbf{U}_{i-1}) - \frac{1}{S} (\mathbf{F}_{i+1} - 2\mathbf{F}_i + \mathbf{F}_{i-1}) - 2\omega_R (\Psi_i - \Psi_{i-1}) \right), \quad (\text{B.13})$$

i.e.,

$$\frac{\mathbf{U}_i^{n+1} - \mathbf{U}_i^n}{\Delta t} + \frac{\mathbf{F}_{i+1} - \mathbf{F}_{i-1}}{2\Delta x} = \frac{S\Delta x}{2\Delta x^2} (\mathbf{U}_{i+1} - 2\mathbf{U}_i + \mathbf{U}_{i-1}) - \frac{S\omega_R}{\Delta x} (\Psi_i - \Psi_{i-1}). \quad (\text{B.14})$$

For the mass equation the last term reads,

$$\begin{aligned} \Psi_i^{mass} - \Psi_{i-1}^{mass} &= \beta \left(\rho_{i+1} - \rho_i + \frac{p_i - p_{i+1}}{\bar{c}_i^2} \right) - \beta \left(\rho_i - \rho_{i-1} + \frac{p_{i-1} - p_i}{\bar{c}_{i-1}^2} \right), \\ \Psi_i^{mass} - \Psi_{i-1}^{mass} &= \beta \left(\rho_{i+1} - 2\rho_i + \rho_{i-1} - \frac{p_{i+1} - 2p_i + p_{i-1}}{\bar{c}^2} \right), \end{aligned} \quad (\text{B.15})$$

where it is assumed, $\bar{c}_i^2 = \bar{c}_{i-1}^2 = \bar{c}^2$.

In the specific flow configuration under consideration, both pressure and velocity are uniform.

510 The Godunov method for the mass equation becomes,

$$\frac{\rho_i^{n+1} - \rho_i^n}{\Delta t} + \frac{(\rho u)_{i+1} - (\rho u)_{i-1}}{2\Delta x} = \frac{S\Delta x}{2} \frac{\rho_{i+1} - 2\rho_i + \rho_{i-1}}{\Delta x^2} - S\omega_R \Delta x \beta \frac{\rho_{i+1} - 2\rho_i + \rho_{i-1}}{\Delta x^2}. \quad (\text{B.16})$$

The “weight” ω_R becomes,

$$\omega_R = \frac{S - u}{2S} = \frac{1}{2} - \frac{u}{2S}. \quad (\text{B.17})$$

The discrete mass equation becomes,

$$\frac{\rho_i^{n+1} - \rho_i^n}{\Delta t} + \frac{(\rho u)_{i+1} - (\rho u)_{i-1}}{2\Delta x} = \frac{S\Delta x}{2} \left(1 - \beta + \frac{\beta u}{S} \right) \frac{\rho_{i+1} - 2\rho_i + \rho_{i-1}}{\Delta x^2}. \quad (\text{B.18})$$

The viscosity coefficient, in the present flow configuration, is consequently:

$$\nu = \frac{S\Delta x}{2} \left(1 - \beta + \frac{\beta u}{S} \right). \quad (\text{B.19})$$

When $\beta = 1$ then $\nu = \frac{u\Delta x}{2}$ meaning that the viscosity is related to transport only, as expected.
 515 When $\beta = 0$ then $\nu = \frac{S\Delta x}{2}$ meaning that the viscosity is greater, as it is function of the fastest wave speed. The viscosity of the Rusanov scheme is recovered. Therefore, parameter β clearly controls numerical dissipation.

Note that for the RSIR RH variant, the mass jump based on the Rankine Hugoniot relations reads,

$$\Psi_i^{mass} - \Psi_{i-1}^{mass} = \beta \left(\rho_{i+1} \frac{u_{i+1} - S_{R,i+1}}{S_{M,i+1} - S_{R,i+1}} - \rho_i \frac{u_i - S_{L,i}}{S_{M,i} - S_{L,i}} \right) - \beta \left(\rho_i \frac{u_i - S_{R,i}}{S_{M,i} - S_{R,i}} - \rho_{i-1} \frac{u_{i-1} - S_{L,i-1}}{S_{M,i-1} - S_{L,i-1}} \right). \quad (\text{B.20})$$

520 In the present context, $S_L = -S < 0$, $S_R = S > 0$ and $S_M = u > 0$ and the previous relation simplifies to,

$$\Psi_i^{mass} - \Psi_{i-1}^{mass} = \beta (\rho_{i+1} - 2\rho_i + \rho_{i-1}). \quad (\text{B.21})$$

The Godunov method (B.18) is then recovered by introducing Eq. (B.21) into Eq. (B.14).

In this transport context, both RSIR and variant RSIR RH solvers present the same viscosity parameter ν : Eq. (B.19). However, only $\beta = 0$ and $\beta = 1$ are admissible for the RSIR solver
 525 based on thermodynamics whereas its variant based on the Rankine-Hugoniot relations admits $0 \leq \beta \leq 1$.

With the RSIR RH solver, when $\beta = 0.5$ then $\nu = \frac{\Delta x}{4} (S + u)$, the fastest wave speed appears, meaning that the viscosity is large again. When $\beta = 1 - \epsilon$ then $\nu = \frac{\Delta x}{2} (u(1 - \epsilon) + \epsilon S)$ meaning that parameter ϵ in β is amplified by the fastest wave speed in the viscosity coefficient.

530 References

- [1] D. Balsara, A two-dimensional HLLC Riemann solver for conservation laws: Application to Euler and magnetohydrodynamic flows, *Journal of Computational Physics* 231 (22) (2012) 7476–7503.
- [2] S. Gavrilyuk, N. Favrie, R. Saurel, Modelling wave dynamics of compressible elastic materials, *Journal of Computational Physics* 227 (5) (2008) 2941–2969.
 535
- [3] R. Saurel, A. Chinnayya, Q. Carmouze, Modelling compressible dense and dilute two-phase flows, *Physics of Fluids* 29 (6) (2017) 063301.
- [4] I. Peshkov, E. Romenski, F. Fambri, M. Dumbser, A new causal general relativistic formulation for dissipative continuum fluid and solid mechanics and its solution with high-order
 540 ADER schemes, arXiv preprint arXiv:1910.02687 (2019).

- [5] E. Toro, M. Spruce, W. Spears, Restoration of the contact surface in the HLL-Riemann solver, *Shock Waves* 4 (1) (1994) 25–34.
- [6] S. Tokareva, E. Toro, HLLC-type Riemann solver for the Baer–Nunziato equations of compressible two-phase flow, *Journal of Computational Physics* 229 (10) (2010) 3573–3604.
- 545 [7] D. Furfaro, R. Saurel, A simple HLLC-type Riemann solver for compressible non-equilibrium two-phase flows, *Computers & Fluids* 111 (2015) 159–178.
- [8] Q. Carmouze, R. Saurel, A. Chiapolino, E. Lapébie, Riemann solver with internal reconstruction (RSIR) for compressible single-phase and non-equilibrium two-phase flows, *Journal of Computational Physics* 408 (2020) 109176.
- 550 [9] V. Rusanov, The calculation of the interaction of non-stationary shock waves and obstacles, *USSR Computational Mathematics and Mathematical Physics* 1 (2) (1962) 304–320.
- [10] A. Harten, P. Lax, B. van Leer, On Upstream Differencing and Godunov-Type Schemes for Hyperbolic Conservation Laws, *SIAM Review* 25 (1) (1983) 35–61.
- [11] B. Einfeldt, C. Munz, P. Roe, B. Sjögren, On Godunov–Type Methods near Low Densities,
555 *Journal of Computational Physics* 92 (1991) 273–295.
- [12] T. Linde, A practical, general-purpose, two-state HLL Riemann solver for hyperbolic conservation laws, *International Journal for Numerical Methods in Fluids* 40 (3-4) (2002) 391–402.
- [13] M. Dumbser, D. Balsara, A new efficient formulation of the HLLEM Riemann solver for general conservative and non-conservative hyperbolic systems, *Journal of Computational
560 Physics* 304 (2016) 275–319.
- [14] A. Chiapolino, R. Saurel, Numerical investigations of two-phase finger-like instabilities, *Computers & Fluids* (2020) 104585.
- [15] E. Toro, *Riemann solvers and numerical methods for fluid dynamics: A practical introduction*. Third Edition, 2009.
- 565 [16] S. Davis, Simplified second-order Godunov-type methods, *SIAM Journal on Scientific and Statistical Computing* 9 (3) (1988) 445–473.
- [17] F. Fraysse, R. Saurel, Automatic Differentiation using Operator Overloading (ADOO) for implicit resolution of hyperbolic single phase and two-phase flow models, *Journal of Computational Physics* 399 (2019) 108942.
- 570 [18] P. Woodward, P. Colella, The numerical simulation of two-dimensional fluid flow with strong shocks, *Journal of Computational Physics* 54 (1) (1984) 115–173.
- [19] R. Loubère, Validation test case suite for compressible hydrodynamics computation, Unpublished notes, Los Alamos National Laboratory (2005).

- [20] E. Toro, The HLLC Riemann solver: A Review, *Shock Waves* 29 (8) (2019) 1–18.
- 575 [21] S. Godunov, A finite difference scheme for numerical computation of the discontinuous wave solutions of equations of fluid dynamics, *Math. Sb.* 47 (1959) 271–306.
- [22] J. Cocchi, R. Saurel, J. Loraud, Some remarks about the resolution of high velocity flows near low densities, *Shock Waves* 8 (2) (1998) 119–125.
- [23] B. van Leer, Towards the ultimate conservative difference scheme. V. A second-order sequel
580 to Godunov’s method, *Journal of Computational Physics* 32 (1) (1979) 101–136.
- [24] A. Kapila, R. Menikoff, J. Bdzil, S. Son, D. Stewart, Two-phase modeling of deflagration-to-detonation transition in granular materials: Reduced equations, *Physics of Fluids* 13 (10) (2001) 3002–3024.
- [25] R. Saurel, F. Petitpas, R. Berry, Simple and efficient relaxation methods for interfaces
585 separating compressible fluids, cavitating flows and shocks in multiphase mixtures, *Journal of Computational Physics* 228 (5) (2009) 1678–1712.
- [26] R. Saurel, P. Boivin, O. Le Métayer, A general formulation for cavitating, boiling and evaporating flows, *Computers and Fluids* 128 (2016) 53–64.
- [27] R. Abgrall, R. Saurel, Discrete equations for physical and numerical compressible multiphase
590 mixtures, *Journal of Computational Physics* 186 (2) (2003) 361–396.
- [28] M. Baer, J. Nunziato, A two-phase mixture theory for the deflagration-to-detonation transition (DDT) in reactive granular materials, *International Journal of Multiphase Flow* 12 (6) (1986) 861–889.
- [29] R. Saurel, S. Gavrilyuk, F. Renaud, A multiphase model with internal degrees of freedom:
595 Application to shock–bubble interaction, *Journal of Fluid Mechanics* 495 (2003) 283–321.
- [30] H. Guillard, C. Viozat, On the behaviour of upwind schemes in the low Mach number limit, *Computers & Fluids* 28 (1) (1999) 63–86.